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AFFDL-TR-71-5 PART I, VOL II



SUBSONIC UNSTEADY AERODYNAMICS FOR GENERAL CONFIGURATIONS

PART I, VOL II-COMPUTER PROGRAM H7WC

J. P. GIESING

T. P. KALMAN

W. P. RODDEN



TECHNICAL REPORT AFFDL-TR-71-5, PART I, VOL II

NOVEMBER 1971

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AIR FORCE FLIGHT DYNAMICS LABORATORY
AIR FORCE SYSTEMS COMMAND
WRIGHT-PATTERSON AIR FORCE BASE, OHIO

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FOREWORD

This report was prepared by the Douglas Aircraft Company, Long Beach, California, for the Aerospace Dynamics Branch, Vehicle Dynamics Division, Air Force Flight Dynamics Laboratory, Wright-Patterson Air Force Base, Ohio, under Contract F33615-70-C-1167. This research was conducted under Project 1370, "Dynamic Problems in Military Flight Vehicles," and Task 137003, "Prevention of Dynamic Aeroelastic Instabilities in Advanced Military Aircraft." Mr. S. J. Pollock of the Aerospace Dynamics Branch was Task Engineer.

This report consists of two parts with two volumes for each part. This volume, Volume II of Part I is the Computer Program H7WC. Volume I of Part I consists of the method of direct application of nonplanar lifting surface elements. Volume I of Part II, contains a method which uses an image system and an axial singularity system to account for the effects of the bodies and Volume II of Part II is the Computer Program N5KA.

The work reported herein was conducted during the period of December 1969 to September 1970.

The Principal Investigator was Joseph P. Giesing. Mrs. Terez P. Kalman was responsible for implementing the method on the computer. Donald H. Larson aided in this implementation and acted as consultant for computer problems. Dr. William P. Rodden, a McDonnell Douglas Company Consultant, contributed many valuable ideas to the project. Others have made significant contributions to this project including Messrs. D. S. Warren and W. E. Henry.

The report was submitted by the authors in November 1970 for publication as an AFFDL Technical Report.

This technical report has been reviewed and is approved.

Datter J. hypertons WALTER J. MYKYTOW

Asst. for Research & Technology

Vehicle Dynamics Division

ABSTRACT

Two methods of accounting for body-liftin satisface interference in unsteady flow are considered. The first method is described in Part I of this report, while the second will be described in Part II to follow.

The first method is a direct application of nonplanar lifting surface elements to both the lifting surfaces and the body surfaces. The body is treated as an annular wing. This type of idealization must be used with an axial doublet introduced to account for body incidence effects. The undesirable effects of the annular wing representation are then reduced.

The second approach, to be described in Part II, uses an image system and an axial singularity system to account for the effects of the bodies.

This volume contains the computer program and contains the FORTRAN listing. Volume I contains the development of the theory, correlation of theory with experimental data, and the parametric study.

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NOMENCLATURE

A	Reference total area
[A]	Matrix of box areas
a inm	Polynomial mode coefficients for mode i
AIC	Matrix of influence coefficients relating generalized forces to generalized deflections for submodes
С	Chord length
¯	Reference cnord length
C _m	Pitching moment coefficient $\left(\frac{\text{moment}}{\text{q Ac}}\right)$, (+ nose up)
C _n	Yawing moment coefficient $\left(\frac{\text{moment}}{\text{q A s}}\right)$ (+ nese right)
Ci	Rolling moment coefficient (moment) (+ clockwise)
C_{Y}	Force coefficient in y-direction $\left(\frac{Force}{qA}\right)$ (+ out right wing) Force coefficient in z-direction $\left(\frac{Force}{qA}\right)$ (+ vertically up)
c_z	Force coefficient in z-direction $\left(\frac{Force}{q A}\right)$ (+ vertically up)
c m	Local moment coefficient $\left(\frac{\text{moment}}{\text{oc}^2}\right)$
c _n	Local normal force coefficient (normal force qc)
e	Strip semi-width
r	Normalized deflection normal to surface (h/s)
h	Deflection normal to surface
K	Kernel
k _r	Reduced frequency, $\omega \tilde{c}/2U_{\infty}$
L.	Lift
M	Mach number; also moment
Q _{ij}	Generalized force
q	Dynamic pressure
q	Generalized coordinates
R_{o}	Body radius
r	$\int (y-\tau_i)^2 + (z-\xi)^2$

s Semi-span

 $\mathbf{U}_{\mathbf{w}}$ Free-stream velocity

W Unnormalized normalwash

w W/U_w

wr w-wb

x, y, z Coordinates of receving points

 \overline{x} , \overline{y} , \overline{z} Element coordinates of receiving points

x Gust reference axis

α Angle of attack; also a function defined by Equation 2.1-15

 β $\sqrt{1-M^2}$; also control surface deflection

 Γ_g Gust dihedral angle. (Γ_g = 0 if gust velocity is vertical)

γ Dihedral angle

ΔC_p Lifting pressure coefficient $\frac{P_{lower} - P_{upper}}{q}$

δ Symmetry index (right and left symmetry); also tab deflection

€ Ground effect index

λ Wave length

 ξ, η, ζ Coordinates of sending points

 $\bar{\xi}, \bar{\eta}, \bar{\xi}$ Coordinates of sending points in element coordinates

P₂ Density at sea level

σ Lateral coordinate in the plane of the surface

ω Frequency

Subscripts and Superscripts

a body axis

B body

c center

f Body or fuselage

g gust

Deflection mode Pressure mode L. E. Leading edge LL Lower left Lower right LR R Axis about which moments are taken Receiving Sending UL Upper left UR Upper right y-direction y z-direction 1/4 One quarter chord point of a lifting surface box 3/4 Three quarters chord point of a lifting surface box

1.0 INTRODUCTION

Computer program H7WC is the result of modifying and extending program H7WB. Program H7WB generates aerodynamic influence coefficients and gust loads for nonplanar surfaces. The modifications to this program include:

- (1) An extension to accept polynomial mode input. (Subroutine GENQ)
- (2) The addition of a capability to determine generalized forces.

 (Subroutine GENF)
- (3) The addition of a capability to account for body angle of attack and camber effects using an axial pressure doublet system.

 (Subroutines AUGW, TKER)
- (4) A refined version of the Doublet-Lattice Method to properly handle problems involving small vertical gaps between wing and tail surfaces. (Subroutine IDF2)
- (5) The addition of the Quasi-Inverse for so ving large numbers of modes or submodes, and for solving future modes with a minimum of computing effort. (Subroutines QUAS, FUTSOL, CXSS)

Program H7WC can handle almost any type of configuration if it is idealized properly. The axial pressure doublets give the proper slender body flow field for an axially symmetric body of any shape. The body/lifting-surface interference is taken care of using lifting surface panels on the body surface. The interference elements must lie on an idealized body of constant cross sectional shape. The rules that apply to the placement of a grid of boxes on the aircraft lifting surfaces also apply to the interference surfaces on the bodies. These rules are:

(1) Boxes (trapezoidal elements) are arranged in strips parallel to the free-stream. The aspect ratio of such boxes should not be large. For the unsteady case, an aspect ratio of order unity or less is preferred.

- (2) Surface intersections, surface edges, fold lines, and hinge lines should lie on box boundaries.
- (3) For wing-tail type configurations the strip boundaries on the tail must be aligned with those on the wing. Strips should be concentrated near the wing tips, near control surface edges and in any region where the span loading changes rapidly.
- (4) The box length must be small compared to the basic wave length. $\frac{\Delta x}{\lambda} \leq 0.04$

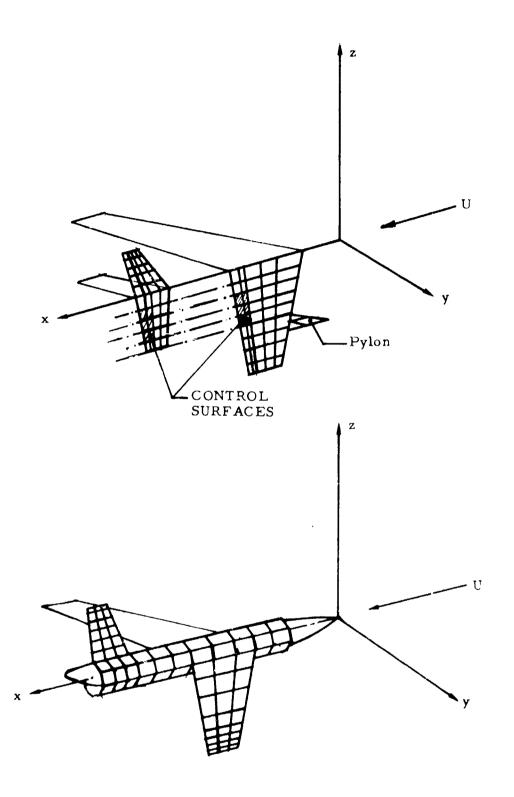
Boxes should also be concentrated near control surface hinge-lines and in all regions where the upwash boundary conditions are discontinuous.

(5) The body cross-sectional shape on which interference lifting surfaces are placed must be constant in the x-direction even for wing-tail problems. (This does not apply to the slender body solution for which the radius distribution is used.)

Sketch 1, 1-1 gives examples of these 5 rules.

Because of the numerical techniques involved, the two-dimensional radial distance, $\sqrt{(y-\eta_e)^2+(z-\zeta_e)^2}$, between a control point, y, z, (3/4-chord) point of a box) and a box edge, η_e , ζ_e , may never approach zero. It does not matter whether the control point is on the same or on a different surface or whether it is up-or downstream from the box in question. This is easily understood for a control point in the wake of the box since the control point may not approach the bound leg of the horseshoe vortex. However, it is also true upstream due to the approach taken for the unsteady effects. (If $k_r = 0$ the upstream effect at $\sqrt{(y-\eta)^2+(z-\zeta)^2} \to 0$ vanishes).

Wing-body-tail problems must be handled using only one interference cross-section. That is, one constant cross-sectional shape must be used both near the wing and near the tail. If this rule is not followed the wakes of the lifting surface element lying on the cross-section near the wing may impinge on the tail surface.



Sketch 1.1-1

When a body mode is input it must be input both to the axial elements and to the interference body panels. It must be remembered, however, that the body is in motion either in the z-or y-directions while the lifting surfaces on the body are in motion normal to themselves. Thus the mode input for a particular body interference panel must be multiplied by $\cos Y$ for (+) z-motion and $-\sin Y$ for (+) y-motion where Y is the dihedral angle of the panel.

The following list gives the program limits.

- 1. The maximum number of unknowns, i.e., the aerodynamic boxes on the lifting surfaces plus the number of body elements, is 400.
- 2. The maximum number of aerodynamic degrees of freedom, in case of AIC calculations, is 130.
- 3. The maximum number of degrees of freedom per strip in case of AIC calculations is restricted to 4.*
- 4. The maximum number of modal right-hand sides is 40 per case.
- 5. The maximum number of aerodynamic boxes per spanwise division (strip) is 30.
- 6. The maximum number of spanwise divisions (strips) is 50.
- 7. The maximum number of bodies is 20.

Program H7WC was used successfully on the following computers: GE635, IBM 7094 and IBM 360/65. Core requirement for the program on the various computers is as follows:

GE635	<u>IBM 7094</u>	IBM 360/65	CDC 6600
28K word	s 32K words	126K bytes	100K bytes

If greater core capacity is available it is desirable to use it up to 260K bytes. The CDC version is slightly modified in dimensions.

^{*} This restriction comes from operational requirements on the IBM 7094. The number of degrees of freedom can easily be increased to the 7 discussed in Appendix D in Part I, Vol I by increasing the dimension 4 to 7 in subroutines AIC. AERO, and FINAL.

2.0 COMPUTER PROGRAM H7WC

2.1 Input Procedure

2. l.l Input Sheets

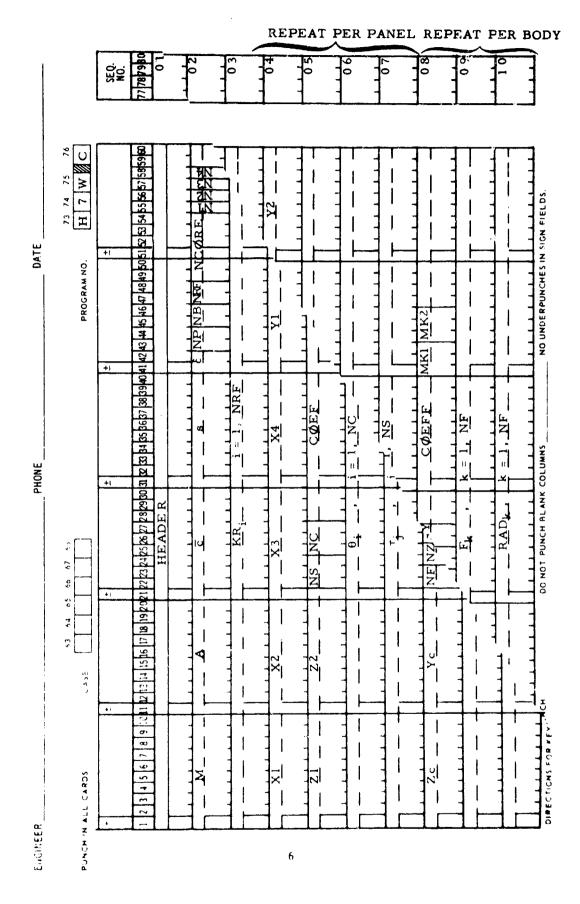
The input sheets for program H7WC are shown on the next three pages. The first three cards represent general case data that is input once per case. The next four* cards (4,5,6,7) represent panel data that is repeated per panel. The panel data must be input in a certain order. The panel data for all "regular" lifting surfaces, (wing, tail, etc.) are input first. The panel data for all interference lifting surfaces (used for the bodies) is input last.

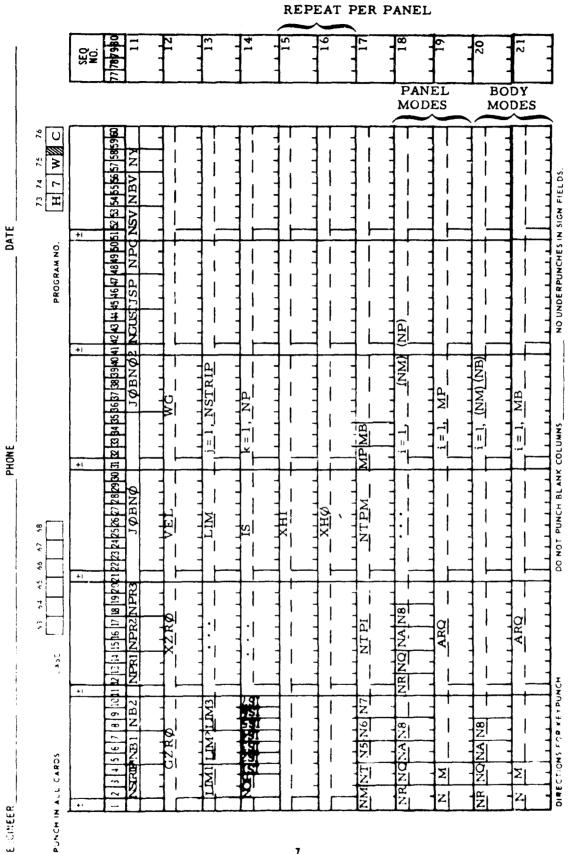
If a body can oscillate in bon the z- and y-directions (nacelle) then it must be replaced by wo bodies: one that moves in the z-direction and one that moves in the y-direction. The subsequent three** cards (8, 9.10) are body data necessary for the slender body analysis. This data is repeated er body. Card 11 is used for miscellaneous modal and AIC data and must be completed in eather case. Card 12 is used when a gust s desired. Card 13 indicates a series of cards that are used to identify lifting surface boxes with strips. Card 14 indicates one or more cards that identify the degrees of freedom to be used for each panel. The card is necessary only if AIC's are desired.

Cards 15 and 16 are used to identify axes for the AIC degrees of freedom. Card 17 s used for modal control data. Card 18 is used to input control data for panel modes, while card 19 represents a series of cards, repeater as necessary, to indicate polynomial mode coefficients for panels. (18ds 20 and 21 are similar to cards 18, 19 except the modes pertain to the z and y modes of the bodies.

where may be more than four cards depending on the number necessary to present all θ_i and τ_i values.

^{*} There may be more than three cards depending on the number necessary to present F and RAD.





086487 ~ XX 20.05 1~ **₹**~ 72 **1**~ 1 76 C 73 74 75 H 7 W NO UNDERPUNCHES IN SIGN FIELDS. 1 DATE PROGRAM NO. 1 -1 NBEL I NBEL LETT NBOX NBOX I XQBN, T = T = 1 NBEL i=1 NBEL DO NOT PUNCH BLANK COLUMNS PHONE 4 1 4 1 8 -- 4 3 8 -- Li IHO DH 2 1 + Li LHO 비 피 H 89 63 3 65 7 CASE DIRECTIONS FOR KEYPUNCH PUNCH IN ALL CARDS NBOX ENGINEER Ą

2. 1.2 Description of Input Data

Item No.

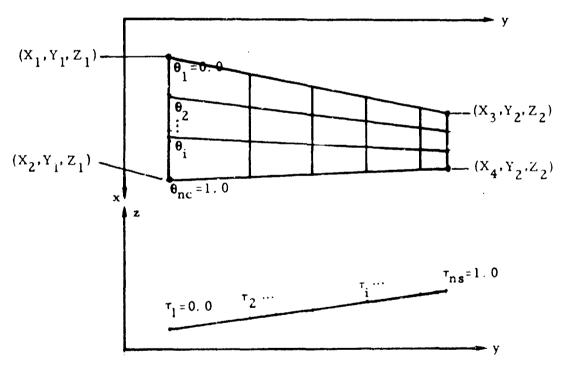
Description

- The header card is used to describe the current run being made.

 Any alphanumeric information can be used, placed in cc 1-60.

 The first 12 card columns should contain the tape title for the AIC's, in the case when AIC's are generated and are to be saved on master tape for later use.
- 2 M, FMACH Mach number, usual definition.
- 3 A, ACAP Reference area; usually total area of wing
- 4 c, REFCHD Reference chord, usually average chord of wing
- 5 s, REFSPN Reference semispan.
- δ, NDELT Symmetry flag (δ = 1 sym., δ = -1, antisym. δ = 0, no sym.).
- 7 NP, NOPAN Total number of panels on all lifting surfaces
- 8 <u>NB</u> Total number of bodies.
- 9 <u>NRF</u> Number of reduced frequencies for which the analysis has to be performed (repeated) recommended maximum is 6 per case.
- NCORE Size of total solution, i.e., NCORE N x M where
 N is the total number of elements or unknowns
 (including z and y fuselage elements) and M is the
 number of modes or the order of the AIC matrix if
 obtained.
- 11 N! If "1", program assumes that case has been run previously and that a quasi-inverse exists on a tape (identified by NTPI) and only new modes are being run 11 "0", program runs a new case.

Item No.	Description
12	N2 - If "0", polynomial modes are to be input. If "1" AIC's are to be calculated.
13	N3 - Data flag. If "1", matrix is printed; if "0", it is not printed.
14	N4 - Detail print flag for slender body solution; 1 to print, 0 otherwise.
15	\underline{KR} - Array of reduced frequency, $KR = \frac{\omega \overline{c}}{2 U_{\infty}}$.
19, 20, 21	X1 X2 X3 - Panel edge coordinates. X4 Y1 Y2 X1, Y1, Z1 (Inboard leading edge) Z1 Z2 X2, Y1, Z1 (Inboard trailing edge) X3, Y2, Z2 (Outboard leading edge) X4, Y2, Z2 (Outboard trailing edge)
24	NS - Number of spanwise divisions for panel considered.
25	NC - Number of chordwise divisions for panel considered.



SKETCH 2.1.2-1

Item No. Description COEF - Scale factor for panel deflection modes. If no scale 26 factor is desired, must be set to 1.0. 27 θ_i - Chordwise divisions in fraction of chord. Usually varies from 0 at leading edge to 1.0 at the trailing edge. T, - Spanwise divisions in fraction of panel span. Usually 28 0.0 at inboard chord and 1.0 at outboard chord. Repeat all panel data per panel (Items 16 through 28) Panels are input in the following sequence: wing, tail, etc., panels, then body interference panels. $Z_c - Z$ -coordinate of body axis. 29 Y - Y-coordinate of body axis. 30 NF - Number of divisions on body. 31 32 NZ - If "1", the program will allow for doublets in the Z-direction and thus will be able to account for Z-or upwash. If not, set NZ = 0. 33 NY - If "1", program will allow for Y-doublets and thus Y-or sidewash. 34 COEF - Scale factor for body deflection mode. MK1- Each body has associated with it interference lifting sur-35 faces. MK1 represents the first box on these surfaces 36 MK2- The last box on the body interference lifting surface panels. Interference lifting surface panels are input after the regular lifting surfaces are input.

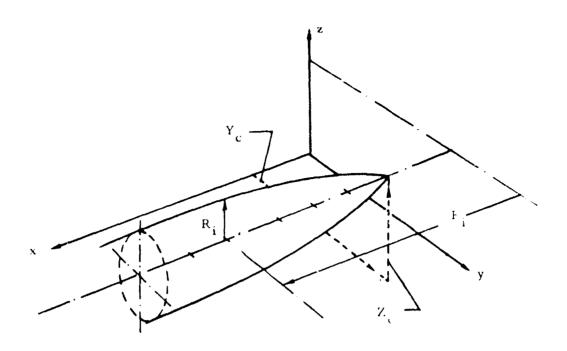
Repeat all body data per body (Items 29 through 38)

RAD - Radius at endpoints.

37

38

 $\underline{\mathbf{F}} = -\mathbf{E}$ ndpoints of body divisions (starts at leading edge and ends at trailing edge). For coordinates of body element end points.



SYETCH 2. 1. 2-2

Item

No.	Description
3 9	NSTRIP - Total number of strips on all lifting surfaces.
40	NB1 - Sequence number of first strip Omit if no AIC's are desired.
41	NB2 - Sequence number of last strip. Omit if no AIC's are desired.
42	NPR1 - Print flag. If "1", print all AIC solutions. Otherwise use "0" (usually = 0).
43	NPR2 - Print flag. If "1", print AIC's. Otherwise use "0" (usually 0)
44	NPR3 - Stability derivative flag. If "I", print static stability derivatives. If "3", print static-and dynamic stability derivatives. Otherwise use "0". Note that, if NPR3 3, NRF must be 2.2 and ER(I) must be 0.0. Usual setting is NPR3=1.
45	<u>JOBNO</u> ~ AIC tape identification. Ignore if no AlC's
46	JOBNO2 - Not used

47 NGUST - Gust flag. Use I for gust, 0 otherwise

item No.

Description

- 48 JSP Second symmetry plane (the plane Z = 0).

 -1 ground effect (antisymmetric)

 1 biplane or jet (symmetric)

 0 no symmetry
- MPC Mode selector for AIC generation.

 Ignore if no AIC's. Use "0" if alternative #1 is used i.e. plunging, pitching, control surface rotation and tab rotation. Use "1" if alternative #2 is used, i.e., three cambering modes plus control surface and tab rotation.
- NSV Total number of strips on all vertical panels which lie on the symmetry plane Y = 0.

 Vertical panels lying on Y = 0 must be input before the other panels. For example the vertical portion of the T-tail must be input first.
- 51 NBV Total number of boxes on vertical panels.
- 52 <u>NYAW</u> Stability derivative override flag. If "1", yaw coefficients will be calculated, if "0", pitch coefficients will be calculated.

If NDELT = 1, NYAW = 0

NDELT = -1, NYAW = 1

NDELT = 0, NYAW = 0 or 1

- 53 GZRO Gust reference plane dihedral in degrees. For 0 dihedral (GZRO = 0) the gust velocities are vertical (in the Z-direction). Omit if gust is not desired.
- 54 XZRO Gust reference point. Point at which sinusoidal gust velocity is unity. Omit if gust is not desired.

Item No. Description **5**5 VEL - Aircraft /elocity in. / sec. Omit if gust is not desired. 56 WG - Gust vertical velocity at XZRO, in. / sec. Omit if gust is not desired. Omit the gust card if the gust is not desired. (Items 53 through 56) 57 LIM - Summation limits for chordwise integration. Usually LIM! is the first box in a strip and LIM2 is the last box where all boxes are numbered consecutively. 58 NOP - Panel number. 59, 60, 61 AIC degree-of-freedom selector. If alternative IS1 IS2 IS3 " 62,63,64 IS4 IS5 IS6 #1 is used (i.e. if NPC = 0) then IS1 through IS6 65 IS7 (IS7 not used) represent: plunging, pitching, control surface rotation, tab rotation, control surface plunging, tab plunging, respectively. If Alternative #2 is used then IS1 through IS7 represent: three cambering modes on the main surface, control surface rotation, tab rotation. control surface plunging, and tab plunging, respectively. 66 XHI - Reference axes measured in fractions of local chord on the inboard edge of a panel. (Further description

- following XHO.) Ignore if AIC's are not needed.
- 67 XHO - Reference axes measured in fractions of local chord on the outboard edge of a panel. For Alternative #1 5 quantities are needed: the elastic axis, the leading edge of the control surface, the rotation point of the control surface, the tab leading edge and rotation point. respectively. The data pair XHI and XHO are repeated for each panel.

Item No.

Description

Omit data Items #58 through #67 if AIC's are not required.

- 68 NM, NMD Total number of modes, i.e., max NQ.
- 69 NT, NTA Total number of ARQ values.
- 70 N5 If "1", save the quasi-inverse on tape so that future modes may be considered. Otherwise set to "0"
- 71 N6 Type of mode input flag
 0 polynomial
 - l input h, dh/dx from tape
 - 2 input h, dh/dx from cards
- 72 N7 If "1", calculate pressure forces and moments.

 Use "0" otherwise.
- 73 NTPI -Quasi Inverse tape identification. Written on tape when Quasi Inverse is formed and used to identify the Quasi Inverse when it is retrieved for additional mode cases.
- 74 NTPM Identification of input mode (h, dh/dx) tape.

The input to follow is associated with the input modes. Equations will be written to define the variables. For panels (lifting surfaces) the deflections will be normal to the surface while for bodies the deflections will be in the Z- (if any) and Y- (if any) directions. For the "NF" panel or body the deflections in the "NQ" mode are calculated as follows:

(f)
$$\frac{(NR)}{NQ} = COEF^{(NR)} \sum_{N=0,1,\dots,M=0,1,\dots} \frac{\sum_{M=0,1,\dots,M} (\frac{x}{s})^{N} (\frac{r}{s})^{M}}{(\frac{r}{s})^{M}} ARQ_{(NR,NQ,N,M)}$$

where r is in a radial direction. The origin of the radius is either at the

origin of coordinates, when N8 = 0, or at the inboard edge (for the panel) or the axis (for a body) if N8 is set to 1.*

$$\tau = \sqrt{(y - (N8)Y_1^{(NR)})^2 + (z - (N8)Z_1^{(NR)})^2}$$
 for panels

$$\tau = \sqrt{(y - (N8)Y_c^{(NR)})^2 + (z - (N8)Z_c^{(NR)})^2}$$
 for bodies

$$\tau = \sqrt{(y - (N8)Y_c^{(NR)})^2 + (z - (N8)Z_c^{(NR)})^2}$$
 for bodies

$$\tau = \sqrt{(y - (N8)Y_c^{(NR)})^2 + (z - (N8)Z_c^{(NR)})^2}$$
 for bodies

$$\tau = \sqrt{(y - (N8)Y_c^{(NR)})^2 + (z - (N8)Z_c^{(NR)})^2}$$
 for bodies

$$\tau = \sqrt{(y - (N8)Y_c^{(NR)})^2 + (z - (N8)Z_c^{(NR)})^2}$$
 for bodies

SKETCH 2.1.2-3

^{*} Since each panel is planar, a is a spanwise distance in the plane measured from its inboard edge or from the origin v and a listance only in the sense that each panel may have a different dihedral. The use of this radial distance is not meant to imply that the panels are curved.

Item No.	Description
75	MP, NMTP - Total number of ARQ coefficients for panels only.
76	MB, NMTB - Total number of ARQ coefficients for bodies.
77	NR - Panel number and/or body number.
78	NQ - Mode number.
79	NA, NARQ - Number of terms (ARQ) in mode NQ, panel and/or body NR.
80	N8 - Flag that sets origin of radial variable . If "0", the x-axis is used. If"1", inboard edge or axis of body is used as origin (see equation for τ).
	The four items NR, NQ, NA, N8 are repeated for each panel or body until all panels and bodies are mentioned. Even if a panel or body is not in motion (NA = 0) the set of four items must be given.
	The direction of motion of a body is given by the NZ or NY flags. Thus if body pitching is given and NY is flagged, then the body will be given a yaw. If a body can move in both the Z and Y directions (a nacelle) then it must be replaced with two bodies; one with NZ = 1 and one with NY = 1.
81	N. LARQ(1) - Power of x/s in the mode polynomial.
82	M, LARQ(2) - Power of r/s in the mode polynomial.
83	\overline{ARQ} - Coefficient of $(x/s)^N (\tau/s)^M$ in the mode polynomial.

Omit modal data cards if modes are not desired (Items 68 through 83).

If modes are input using the options N6=1, or N6=2, omit !tems 73 through 83, and add the following data items:

Item No.			
84	NBØX	-	Total number of boxes on all panels.
85	<u>B</u>	-	Generalized force integration matrix (box area) X (displacement 1/4-chord of box).
86	<u>H</u>	-	Displacement of 3/4-chord point normalized by semispan, h/s.
87	DH1	-	Derivative of normalized displacement by (x/s) at $3/4$ -chord point of box $d(h/s)/d(x/s)$.
88	NBEL	-	Total number of body elements.
89	B	-	Generalized force integration matrix (body element length) X (diameter) X (displacement 1/2 -chord of element).
90	<u>H</u>	-	Displacement of 1/2-chord point normalized by semispan, h/s.
91	DHI	-	Derivative of normalized displacement by (x/s) at $1/2$ -chord point of element. $d(h/s)/d(x/s)$.
92	DH2	-	$d^{2}(h/s)/d(x/s)^{2}.$

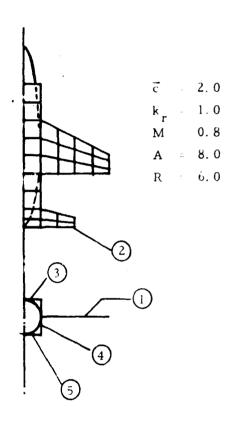
NOTE: If N6=1, Items:#84 through #92 are input from cards according to the formats shown on page 8.

If N6-2, Items #84 through #92 are read from tape unit MTAPE, as one record per variable, NBØX, $(B_i, i=1, NBØX)$, $(H_i, i=1, NBØX)$ and $(DHl_i, i=1, NBØX)$. If bodies are also present, five more variables are read from tape unit MTAPE; again, as one record per variable: NBEL, $(B_i, i=1, NBEL)$, $(H_i, i=1, NBEL)$, and $(DH2_i, i=1, NBEL)$.

For each additional mode, repeat Items #85 through #92.

2.2 Example Case

The following case is to be viewed only as an example of the proper input procedure and not as an example of an ideal or optimum idealization. The configuration is shown in Sketch 2.2-1.



Sketch 2.2-1

The fuselage is divided up into 10 equal elements for the slender body calculation. The divisions along with the radius is given as follows:

$$\mathbf{F}_{\mathbf{k}} = 0, 1, 0, 2, 0, 3, 0, 4, 0, 5, 0, 6, 0, 7, 0, 8, 0, 9, 0, 10, 0$$

 $RAD_{\mathbf{k}} \simeq -0$, 0.7, 0.8, 0.9, 0.95, 1.0, 0.9, 0.8, 0.7, 0.5, 0.0

The axis of the fuselage lies on the x-axis, $Z_{\frac{1}{C}} = Y_{\frac{1}{C}} = 0$ and it is assumed that the fuselage moves only in the z-direction (NZ=1). The modes of

oscillation are symmetric (δ = 1.0) pitching about the middle of the fuselage and plunging

$$h/s = -5.0 + 5 x/s$$
 (where $s = 5$)
 $h/s = 1.0$

These modes apply to the entire configuration. The panels are numbered from 1 to 5 in Sketch 2.2-1.

The panels are given as follows:

(1)
$$X_1 = 4.0$$
, $X_2 = 7.0$, $X_3 = 6.0$, $X_4 = 7.0$
 $Y_1 = 1.0$, $Y_2 = 5.0$, $Z_1 = 0$, $Z_2 = 0$
 $\theta = 0$, 0.333, 0.666, 1.0, $\tau = 0$, 0.25, 0.50, 0.75, 1.0

(2)
$$X_1 = 9$$
, $X_2 = 10$, $X_3 = 9.5$, $X_4 = 10.0$
 $Y_1 = 1.0$, $Y_2 = 3.0$, $Z_1 = 0$, $Z_2 = 0$
 $\theta = 0, 0.5, 1.0$
 $\tau = 0, 0.5, 1.0$

(3)
$$X_1 = 2.0$$
, $X_2 = 10.0$, $X_3 = 2.0$, $X_4 = 10.0$
 $Y_1 = 0$, $Y_2 = 1.0$, $Z_1 = 1.0$, $Z_2 = 1.0$
 $\theta = 0$, 0.125, 0.25, 0.375, 0.50, 0.625, 0.75, 0.875, 0.9375. 1.0

(4)
$$X_1, X_2, X_3, X_4, \theta \text{ same as } (3)$$

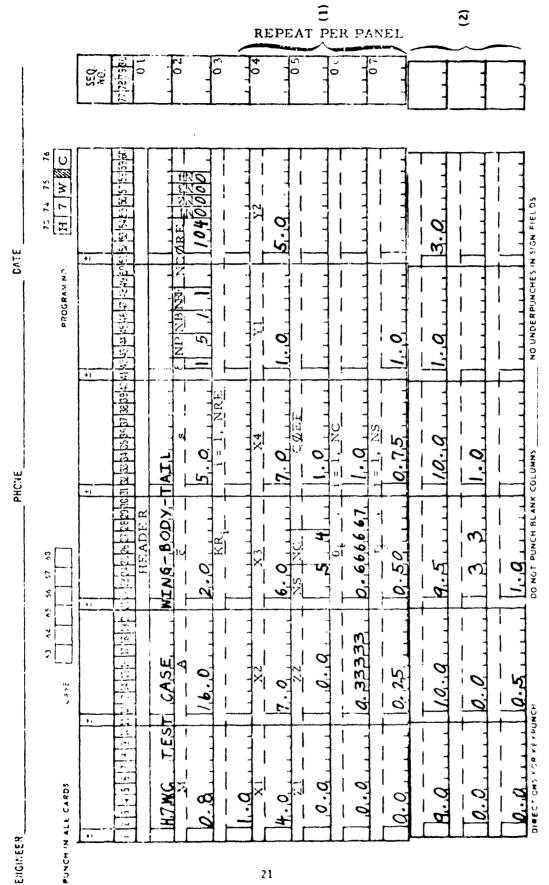
 $Y_1 = 1.0, Y_2 = 1.0, Z_1 = 1.0, Z_2 = -1.0$
 $\tau = 0, 0.5, 1.0$

(5)
$$X_1, X_2, X_3, X_4, \theta, \tau \text{ same as } (3)$$

 $Y_1 = 1, 0, Y_2 = 0, 0, Z_1 = -1, Z_2 = -1$

The input sheets are shown on the next several pages. The output is shown on the subsequen; pages.

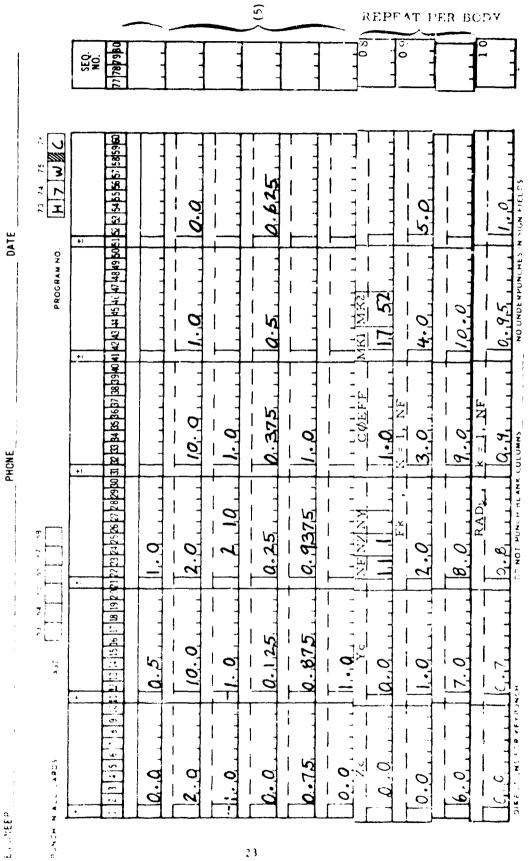
2. 2. 1 Input Sheets



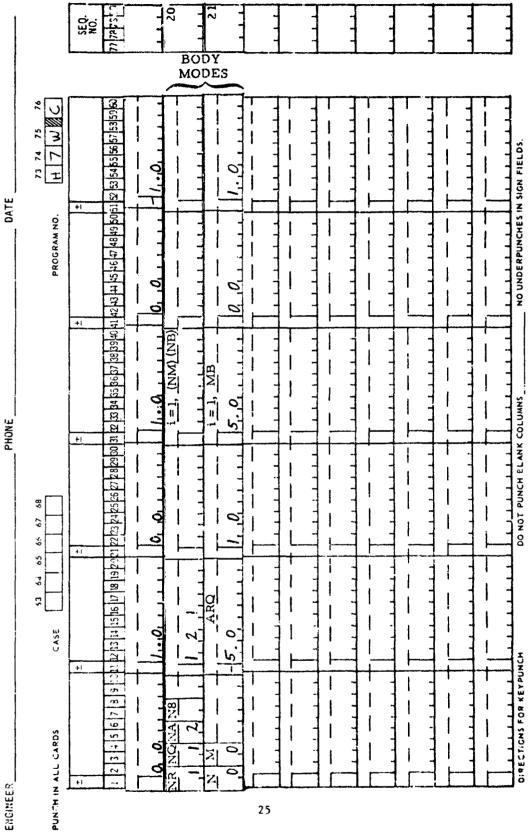
787980 SEO NO. H 7 W C 0.625 .0 NOTINGE RPUNCHES IN SIGN FIELDS DATE PROGRAM NO. 1 6.5 1.0 11:19 10.01 9.375 0.375 DO NOT PUNCH BLANK COLUMNS PHONE 9.9375 9.937.5 2.0 3,10 0.25 0.25 ıč. [9.875 0.125 | 0.875 0.0 304 Ö HUNCHEN AUGUSTAC . LESSO 0.0 ١ 0.75 0.75 SCEND A M. HONDS 7.0 0 0 8334713 22

(3)

<u>4</u>



		SEQ. NO.	77 787980]		13			-		PANI	EL ES	}	<u></u>	1		
	13 14 73 16 [H] 7 [W] C]	,	096989129 9 5299189185		NSV NBV NY	1 \sqcap	1,1,1,1	111111		1			النييييييا		2000	50	7.00
DATE	YROGRAM NO.	→ 1	. इति हम		NGUSTUSP NPC N] [1,13,14	4 1 1 1 1 1 1 1		+++++++++++++++++++++++++++++++++++++++	10,10				10.00], .0, .1	0.0
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ď	b, 20 00		11 72 (21 (21 (22 (22 (22 (22 (22 (22 (22 (2	0,.7,0,,,,,	QNBQL	WIT -	P1-14	35 43	- Mark -		3 1 0 1		11, 10, 12, 14	 	11,0,1	Tc . c	1
- militarite in internetamente mann a same matthespelei	13 64 65			0,.80,	NERI NERZNERS	N - N - 1	9, 1	26 34	$ \frac{1}{10}$ $\frac{1}{10}$	1	2 - 2 - 2		3 2 0 1		-5.0	5.5	5.0
	80 0 4 4 0 0 0 0 4 X	•		0.90	TRUMPINE IN BE	13	ന_	27 71	Tex Tax land and level	0 5			2 2 1 1		10 0		00
833.1888	X V Z							24									•



2.2.2 Out put

MINE FEST CASE MINE-BOOT-TAIL

.. ARRAY OF REDUCED PREQUENCIES ..

4 CHEROMESE, DIVESTORS FOR PAREL 1

0.0 0.333329906 00 0.64666488 00 0.100000000 01 5 SPANNISE DIVISIONS FOR PANEL 1 0.0 0.25000000€ 00 0.500030000 00 0.75000000E 00 0.100000006 01 20 ... XI ELEMENTS FOR PANEL 1000 0.400000000 01 0.499998956 01 10 301000604.0 9.70000000E 01 0.450000000 01 0.511112646 01 0.414444796 01 0.70000000 01 5.9000000F 01 0.566665946 01 0.63133340E 01 0.700000006 61 0.550000000 01 0.59999943E 01 0.45000000E n) 0.70000000F 01 0.400000000 01 0.433332926 01 0.666665 '08 01 0.700000006 01 20 *** ETA ELEMENTS FOR PANEL 1*** 10 30000000F.01 0.100000006 01 0.1300000000 01 0.10000000E OI 0.200000006 01 0.200000006-01 0.20000000€ 01 0.20000000€ 01 0.300007U0E 01 0.100000006 01 0.100000000 01 0.30000000 01 0.400000000 01 0.400000000 01 0.4000000000 01 0.40000000E 01 0.500000000 01 0.500000006-01 0.50000000F 01 0.500000006 01 20 ZEE ELEMENTS FOR PANEL NO. 1 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.6 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 4 PPP RI-J ELEMENTS FOR PANEL 1000 0.425000000 01 0.47500000E 01 0.525000006 01 0.575000006 01 C.HEGGLES FOR PANEL NO. 1 0.275000000 01 0.224999906 01 0.17500000F 01 0-125000006 01 ... PANEL HO. 2 INPUT VALUES ... 9.000000 XZ = 10.000000 Y1 = 1.000000 Z1 = 9.500000 N4 + 10.000000 V2 + 3.000000 Z2 +

3 CHURDWISE DIVISIONS FOR PANEL 2

0.10300000E 01

NC = 3 NS = 3 NOELT = 1 NO. OF PANELS = 5

0.500000000 00

0.425000000 00

#1 * 2.000000 #2 * 10.000000 *1 * 0.0

*** PANEL NO. 3 IMPUT VALUES ***

0.0

0.475000000 00

•	3 SPANHESE DEV	ISIONS FOR PANEL	2		
0.0	0.50000000000000	0.10000none n1			
	•	XI ELEMENTS	FOR PANEL 2000		
10 3000000000	0.95000000E 01	0.10000000E 02	0.925000000 01	0.942500006 01	0.10000000 02
0.95010000E GL	0.475000000 01	9.10000000E 92			
	•	*** ETA ELEMENTS	FOR PANEL 2000		
1.10000000 01	0.1000000000 01	0.100000000 01	0.2000000ut 01	0.200000005 01	0.20000000E 01
0.1000000000 01	0.30000000C OL	0.300000000 01			
		• ZEE ELEMFHTS	FUR PANFE NO. 2		
0.9	0.0	0.0	0.0	0.0	0.0
0.7	0.0	0.0			
	2	*** 11-J ELEMENT	S FOR PANCE 2000		
0.91250000€ 01	0.41750000F 01				
	C.WIGGLES FOR PANEL	40. 2 *****			

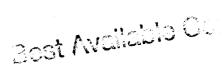
21 .

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1.000000

TO CHANDRIZE DIAISTONS FOR PAREL 3

	to eminomise t	STATESTONE SOR SWEET	1		
0.0	0.125000000 00	7. 254000001 00	0.31500000€ 00		
0. 150000000 00	0.#f500000F 30	0.41710000 00	0.19000000 01	0.100000000000000	0.675000006 00
	P SPANUTSE U	IVISIONS FOR PANEL			
U, U	0.1000000000 01				
		SO *** HI EFENEME.	S FOR PANEL 3000		
3.2000000000000001	0.3000000ur ul	0.40/10/10/00 01	0.900000000 01	0.60000000 01	
9.4000000000000001	0.400000000 01	0.950000000 01	0.10000000000000	0.200000000 01	0.100000001.0
0.400000000 01	U.50000000t 01	0.60000000000001	0.700000000 01	0.#000000000	0.900000006 01
0.9500000000001	0.100000000 02				2777777777777777
		30 ELV EFENEMA	S FUR PANEL 3000		
0.0	0.0	0.0	0.0	0.0	0.0
0.100000006 01	0.0	3.0	0.0	0.100000000 01	0.100000000 01
0.1000000e nt	0.10000000000000	0.10000000 01	0.1000000000000001	0.100000000 01	0.1000000000000000000000000000000000000
	••	20 EFF LLEMENTS	FOR PANEL NO. 3		
0.1000000000001	0.100000099E 01	0.10000000f of	0.100000006 01	0.10000000E 01	1.199000016 01
0.100000000 01	0.1000000000001	0.100000 006 01	0.100000000 01	0.100000000 01	0.100000000000001
0.100000000 01	0.100000000 01	0.1000000000 01	0.19000000+ 01	0.100000001 01	9.100000000 01
f.innonconf ot	0.100000000				
		1 *** XT-J ELEMENT	S FUR PANEL 3000		
0.20000000E 01					
0.#0000000 01	C.WIGGLES FIJH PANE	L 1917. 3 *****			
010300.00	*** PANEL NI). 4	INPUT VALUES ***			
#1 • 2.000unn	#2 = 10.000300 YE		• 1.000000		
E3 4 2.000000		• 1.000000 22			
NC + 17 NS + 3	NOELT . 1 NO. OF P				
	TO CHOHOWESE DE	VISIONS FOR PANEL	•		
0.0	0.12500000F 00	J.250000J0F 00	0.37500000€ 00	0.573000001 00	7-62500000€ 00
0.750000006 00	0.4750000ne 00	0.937500006 00	0.100000000 01		700000
	3 SPANHISE DI	VESTONS FOR PAYEL	•		
1.0	0.50000000F 00	0.10000000 01			
	1	O *** XI ELEMENTS	FOR PANEL 4000		
1.200000000 01	0.100000000000	0.400000000 01	0.500000000 01	0.600000000 01	0.700000000 01
0.9000000F 01	0.97070300F 01	0.9500000F 01	0.10000000F 02	0.2000000000 01	0.3000000000 01
0.400000001 01	0.500000000 01	1.600000000 01	0.700000001 01	0.807000000 01	0.400000000 01
9.950000006 01	0.10000000F 32	10 30000000000000	0.30000000 01	0.40000000F OI	0.500000016 01
0.60000000F AL	0.70000000000	0.8000000000 01	0.900000000 01	0.95000000E 01	0.100000000 02
U.10000000 01	3.1000000E-01	D ETA ELEMENTS			
0.1000000000 01	0.109000000 01	0.100000000 01	0.1000000000001	0.10000000F 01	4.10000000E 01
0.100000000 01	0.100000000 01	0.100000000 01	0.100000001 01	0.100000000 01	0.100000006 01
1.199909996 OI	0.100000000 01	7-10/10/10/10/1	0.10000000 01	0.10000000E 01	0.100000001 01
0.100000000001	0.100000006 01	0.100000000001	0.100000000 01	0.100000000 01	0.100000000 01
		30 ZEE ELEMENTS			01140400000 (1)
0.10000000E 01	0.100000000 01	0.10000001.01	0.100000000 01	0.1000000001 01	0.100000000 01
0.100000000 01	0.10000000000001	0.10000000E 01	0.100000000 01	0.0	0.0
0.0	9.0	0.0	0.0	0.0	n.a
a.10000000F A1	0.0 -0.10000000000000	-0.100000000000000000000000000000000000	-0-103000300 01	-0.1000000000 01	-0.100000000 01
5.15.15.1.10TF 111	-vetoowingt of	.u.100000000 ol	-0.100000000 01	-0.100000001 01	-0.10000000 01
9.20000000 01	0.20000000 01	HI-J 16141413	- 14 - 446 F # 4 - 4 - 4 - 4 - 4 - 4 - 4 - 4 - 4 - 4		
· · ·	CHILGGERS FOR PANEL	NO. 4 ****			
0.80000000€ 01	0.80000000F 71				
	*** PANFL NO. 9	IMPUT VALUES ***	27		
			41		A - mo i



```
TO CHORONISE DEVISIONS FOR PAREL S
                      0.12500 1006 10
                                          9.250000000 00
                                                               U. 375000000 00
                                                                                   U.$4000000F DO
                                                                                                        U.425000001 DO
 0.140000000 00
                      3.479000000 00
                                          0.43750000+ 00
                                                               0.100000000 01
                                      20 *** XI ELEMENTS FOR PANEL 5***
 d.2030aonae al
                      0.1000000000
                                          0.400000006 01
                                                               0.500000000 01
                                                                                    0.600000000 01
                                                                                                        0.7000000t 01
 4.40000001F 01
                      0.4000000F UI
                                          2.950030001 01
                                                               0.100000000 02
                                                                                    0.200000306 01
                                                                                                        0.100000000 01
 0.40000000000001
                      0.500000000 01
                                          10 300000000
                                                               0.700000000 01
                                                                                    10 300000008.01
                                                                                                        0.400000004 01
                      0.100000000 02
                                      20 *** ETA ELEMENTS FOR PANEL 5000
 a. Laguerance of
                      n. Innoquone oi
                                          0.100000000 01
                                                               0.100000000 01
                                                                                    0.106000006 01
                                                                                                        U.100000000 01
 0.130903996 01
                      0.100300306 01
                                          0.100000000 01
                                                               0.1000000001 01
                                                                                    n.0
                      0.0
                                           20 ZEE FLEMENTS FOR PANEL NO. 5
•0.104000001 a
                     -0.103300036 01
                                         -0.100000001 01
                                                              -0.100000000 01
                                                                                   -9.100000001 01
                                                                                                        -0.100000000 01
-0.10000000E 01
                     -0.10000000F 01
                                         -9.190093006 91
                                                              -0.10300000F 01
                                                                                   -0.10000000F 01
                                                                                                        -0.10000000F 01
 -0.10000000F 01
                     -0.1000a000F 01
                                         -0.10000100F 01
                                                              -0.1000000F 01
                                                                                   -0.100000001 01
                                                                                                        -0.1000000000 01
-0.10000000F 01
                     -9.10000uduf 31
                                       1 900 KI-J PLEMENTS FOR PANEL 5000
 0.200000000 01
                     C.WIGHLES FOR PANEL NO. 5 .....
 0.400000001 01
12 16 25
                        5 DIMEDRAL ANGLES FOR ALL PANELS
                                                                                    0.180000000 03
                                            52 *** -K- ELEM. FOR ALL PANELS ***
                      0.58541672E 01
 0.49374905F 01
                                           0.67708171F 01
                                                               0.531749246 01
                                                                                    0.60674967E 01
                                                                                                         0.6512-9916 01
 0.56474924E 01
                      0.627083026 01
                                           0.66541460E 01
                                                                                    0.64791641E 01
                                                                                                         0.66958321F 01
 0.94511250F 01
                      0.9890A250E OL
                                           0.96293750F 01
                                                               0.942187506 01
                                                                                    0.275300000 01
                                                                                                         0.375000000 01
 0.475300ME 01
                      0.575000006 31
                                           0.67570000F OL
                                                               0.775000006 01
                                                                                    0.475000000 01
                                                                                                         0.93750000E 01
 0.447400006 01
                      0.27500000E 21
                                           0.175030306 01
                                                               0.47500000F 01
                                                                                    0.575070006 01
                                                                                                         0.47500000F 01
 9.775000335-01
                      0.875000006-01
                                           9.437500006 01
                                                               0.987500006 01
                                                                                    0.275000008-01
                                                                                                         0.375000000 01
 0.475000000 01
                      0.57500000€ 31
                                           0.67503000F 01
                                                               0.775000001 01
                                                                                    0.475000301-01
                                                                                                         0.93750000F 01
  0.98750000F 01
                      0.27500000F 01
                                           0.175000006-01
                                                               0.475000000 01
                                                                                    0.575000006 01
                                                                                                         0.675000000 01
 0.775000000 01
                      0.87500000F 31
                                           0.93750000F 01
                                                               0.98750000F 01
                                            57 *** KT-V FEFMENTS ALL PANELS ***
                                           0.631750006-01
  7.447916416 01
                      0.534582546 01
                                                               0.49374971F 01
                                                                                    0.544744336 01
                                                                                                         0.64175000F 01
  0.539583116 01
                      0.19/916135 01
                                           0.656250006 01
                                                               Q.58561651E 01
                                                                                    0.6270#2828 01
                                                                                                         D. 66875000F 01
  0.921437501 01
                      0.947147306 01
                                           0.945312306 01
                                                               0.976562506 01
                                                                                    0.225000000 01
                                                                                                         0.125000004 01
  0.425000008 01
                      0.525000000 01
                                           0.62500700E 01
                                                               0.72500000€ 01
                                                                                    0.825000006 01
                                                                                                         U.91250000F 01
  U. 1047U000E 01
                      0.22500000€ 01
                                           0.325000006 01
                                                               0.42500000€ 01
                                                                                    0.525000000 01
                                                                                                         0.023000001 01
  0.72500000F 01
                      0.87500000F 31
                                           0.912500006 01
                                                               0.462500006 01
                                                                                    0.22500000F 01
                                                                                                         0.12500000f 01
                                           0.42500000F 01
                                                               0.725000000 01
                                                                                    0.024000000 01
  0.425000000 01
                      9.525000007 01
                                                                                                         0.912500000 01
                                           0.325000006 01
                                                               0.425000000 01
                                                                                    0.525000006 01
  0.962500006 01
                      0.225000006 01
                                                                                                         0.425000000 01
                                           0.41250000E 01
                                                               0.962500000 01
                      0.823000006 01
  0.72500000# 01
                                            52 *** ET-LELFHENTS ALL PANELS ***
  0.42477971F 01
                      0.57499914E 01
                                           0.424000000 01
                                                                                    0.55416594E 01
                                                                                                         0.43750000F 01
                      0.5#33327# GE
                                           0.650000000 08
                                                                3.55249981E 01
                                                                                    0.612499526 01
                                                                                                         0.462500006 01
  9.51444441F 01
  0.912500005 01
                      0.96250000 01
                                           0.934375006 01
                                                               0.971975006 01
                                                                                    0.22500000F 01
                                                                                                         0.325000000 01
                      0.521000006 21
                                           0.42500000F 01
                                                               U.72500000F 01
                                                                                    0.624000001 01
                                                                                                         0.412400000 01
  0.423000006-01
                      0.22500000F 01
                                           0.325000000 01
                                                               0.423000000 01
                                                                                    U.425000001 01
                                                                                                         0.625000001 01
  3.962500006 01
  0.725000006 01
                      0.825000006 01
                                           0.412500006 01
                                                               0.962500000 01
                                                                                    1.225000001 01
                                                                                                         U.32500000F 01
                                                                                    0.425000000 01
  0.423000000 01
                      0.525000000 01
                                           0.8250000000 01
                                                                0.725000000 01
                                                                                                         0.912500006 01
```

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., ... x1-2 FLEMENTS ALL PANELS ...

0.325000000 01

0.91250000F 01

0.44250000F 01

9. 721000001 01

0.775000000 01

0.475000000 01

0.425000000 01

0.46250000E 01

0.525000006 01

P. +2500000F 01

0.410033111			1005 01 (01	0.545537738	01	0. +1 4000991	0 t
0.762479618				0.406333211	Ó1	955644144.0	01	0.47500000	01
0.111000000				300015141.0	01	0.235000001	01	9.12500000€	01
0.96250606				10000005	01	0.425000001	01	0.41230000f	01
0.11100000E				0.425000001	01	0.52500000F	n:	0.625000000	01
0.+230000000				300006		0.55200,000	-	0.121000006	01
0.742300006				7.725000000		0.024000000		0.41520000f	01
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	***************************************		P -Y- ELEM. FO	9.9e250000€	-				
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0.454250006-00	n	0.781250006	ტი	0.497625006	nn	0.953125006	ne				
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0.100000000	1	o. Langonnage	01	0. E00000000E	91	0.10003010f	01	C. Launnonof	o s	0.100000000	01
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	L I	L KNOY ELEMFNIS FUR K	IODY NO. 1			
		9.20000000000000	0.30000000		a.400000000 01	0.500000001 11
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0.0	0.695999996 73	0.744449956 00	0.849994786	60	0.44494444 00	U.14000000F 01
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		10 A ELEMENTS	FUR ALL BURIES	•		
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		10 ATV ELEMENTS	POR MEE BODIES	•		
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	0.750000000 01	0.450000006 01	0.95000000			
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0.10000700F 01	0.100070000 01	0.100000000 01	0.10000000	()!	0.100000000 01	0.10703009F 01
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0.162399965-00	0.74999952F 03	0.864939906-00	0.92499971	F 07	0.47449496	3.44444IR 00
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	Int 52 8-4	ATRIK FLEMENTS FO	DK MODE NO. 1		
-0.1909//5906-01	9.1451 10536-44	0.401249211-51	-0.147509346-02	U. 706/49171-01	0.471248846-01
0.453544546-03	0.226422506-01	0. 164541 481 -01	7.147338896-01	0.211407276-01	0.281249596-01
0.7410144#E-01	0-41 /5/ //46-01	J. > 70+4703E-01	0.545703136-01	-0.110000011 00	-0.69499941-01
-0. 100000 151-01	0.49499632E-02	0.049797975-01	9.44999148-01	0.1/9949AHE 00	0.#24999215-01
0.426444126-01	u. 0	3,9	0.0	0.0	0.9
0.0	0.4	0.0	0.0	0.0	0.9
1.1	0.0	0.9	J.0	0.0	0.0
2.0	0.11/10/0/018 00	3.64344444	0. 170000331-01	-0.944984726-07	-0.499999911-01
-0.419999166-01	-0.12999986 00	-0.474994516-01	-0.974994176-01		
	THE 32 H-1	INTRIX ELPMENTS FO	34 MODE NO. 1		
0.40000000000101	-0.112572456 00	0.500100006-01	0.42707420E 01	0.500000000 01	0.845-150#6-01
9.5000uoags at	0.196249235 01	9.120303306 01	U.5 11 24666E 01	0.50000000F 01	0.40u24804F 01
0.900000028 21	0.163766756 01	0.501103398 01	0.63561412F 01	n. 20000000F 01	0.92704111 01
a. snogannat at	0.531246466 01	0.300000F 01	G.7395#15#F UE	0.50000000E 01	0.447414175 31
9.900000000 91	0.222556256 02	0.50000000 01	0.24451110£ 02	0.400000001 01	0.23046#75F 02
0.500000001	0.24609175E 12	0.7	0.0	o.n	0.0
0.0	0.3	1.0	0.0	0.0	0.0
0.0	0.0	2.2	1.0	n.a .	0.0
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1.0	0.0	0.3	0.0	0.0	0.0
0.0	0.0	3.0	0.0	0.0	0.7
1.0	0.0	7.7	2.0	0.0	0.0
0.0	7.0	4.4	1.0	0.0	0.0
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	THE 52 8-1	MATRIX ELFMANTS F	38 400t 40. 2	•	
0.306462746-01	7.366670798-01	0.344146396-01	0.274996738-01	a. 10000340f-UL	0.799999797-01
9.211339716-01	0.21331405F-01	3.233333196-01	0.146664686-01	0.16466#h7f-01	0.166666581-01
0.174994947-01	0.176919948-01	7.124949998-01	3.124994996-01	0.39999998-01	n. 344444441 -01
g. 19994999£-01	0.144444446-01	0. 194999991 -01	7.14444446-01	0,399999996-01	0.20000000-01
0.2000000000-01	0.0	0.0	a.n	0.0	0.0
0.0	0.0	0.0	0.0	1.0	2.0
0.0	0.0	0.0	0.0	0.0	0.0
1.0	-0.379999996-31	-7.194343438-01	-0.344949476-01	-0, 199419997-01	-0.399979996-01
-7.3 4444444 F-7	-9.39999499E-01	-0.2400000000-01	-0.20000006-01		
		MATRIX FLUMENTS P			
2.0	U-103000000 01	9.3	0.1000000000 01	0.0	0.50007000F 01
7.0	a.snanusque di	n.a	9,50000000# 01	0.0	0.300030000 01
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n.a	9.500000000 01	0.0	0.1000000000000000	0.0	0.100000000000000
1.6	0.10000000F 01	0.0	0.100000000 01	n.o	0.50000000000001
4.4	n.300000000 01	0.0	3.0	4.0	0.0
4.4	0.1	1.0	0.0	0.0	0.0
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                                                                               0.18999886F-01
  4.509444318-01 0.449464F-01 0.43339457E-01 0.4499961F-01
                      THE 10 AS MATRIA ELEMENTS FOR MODE NO. 1
  0.50000000 01 -0.225000000 02 0.50000000 01 -0.125000000 02 0.50000000 01 -0.125000000 02
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COL. NO. 1 UP DEL-CP MATHER
  9.00/911718 92 -0.979009/38 92 0.003602008 02 0.125003506 07 0.50020906 07 0.18449938C 07
  1.21549749 02 0.76705506 02 3.42246546 01 0.22865108 07 -0.106078781 02 U.74774527E 02
  -n. 251.091.06 02 0.21.391190r 02 -n. 12.991474E 02 0.1.57.39494 02 -0.39499952 02 -0.31.615768F 01
  -0. 13179195E G2 -0.02#1178FE J2
        CHEUMS NO. . E OF THE SUGA MATREA
  0.25395145F 01 -0.50628762E 01 -0.762761552 01 -1.11578839F 01 -0.7795246E 01 -0.88472014F 00
  -0.76691551F 00 -0.267616850 01 -0.1150000000-01 -0.23530371E 01 0.64288786 00 -0.19889160F 01
  -n. ipperson ni -g. nebesi/it og -n. tisioniis oi -o. tissoneini on -o. seenaaan ni -o. in seenaan ni
  0.189125318 01 0.13764852F 01 0.13761346E 01 0.23404674E 01 0.15545197E 01
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d. Hankselv 31	1-244553166 (6)	J. 120#45701 - 01	0.184419741 00	9.115403456	01 0.411141861 00
1,33564544 30	3. 146411711 00	9-951946131-00	-4.1177609Nt QJ	4.655524146	00 -0.716416501 00
a. Myspersi (a)	- 1,403642688 35	O.11 feetent ou	+9.423454 59 80	0.159192666	00 -0.4/11*5/04 10
9 .11111111 00	1,110044396-14	fillseverable on	3.foktonist no	0.675047955	W1 P. 40145671 BU
3,585091108-20	1,156294998 11	New J	9.0	4,4	4.5
1.47	4.3	.))	0,0	0,3	0.0
3. F	2. 1	J. 3	3.3 · · ·	v.0	6.0
0.4	1,1	4.9	9.0	0.0	0,0
9-15	++ 1	2.9	v.u	0.3	0.0
9.0	1.1	1, 2	3.0	ŭ. o	0.0
0.0	1.1	1, 3	9.1	0.0	0.9
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40¢ 43.	T ()f		•	1	BP1 220P12	
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ı	1,04111	***7114	1,50,00	14.3	1. 12 11 21 41 5 - 02	*U.113113661 07
,	3. 48445	5.1 6581	1,40350	9.2	3.615597567 02	4, 114469751 02
•	9,75429	0.11/0)	1.51114	4.4	0.1-1771441 02	u. Funt front O2
•	9,44111	*,41750	2,53356	a,a	1-23-103224 12	-9,14441573[02
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,	2.04111	1, 19541	9.53340	1,1	7.236245006 92	-0.469066456 01
•	1.41000	3. 17916	1,55113	71, 7	7.1477#330F 02	0,724745446 07
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. 10	4.04111	5.85617	6,5 3303	9.0	3+24551270F 02	0.135413556 02
11 -	7.41446	4.27341	4.53330	1.0	0-124704571 02	0.254799678 02
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1.5	0.12570	9,21114	1.50300	0.0	1.111743776 01	0.17185429F 03
1 .	0.57577	V. A714A	1.51110	0.3	-0.154727081 01	0.423662526 02
1.	0.12500	4.45311	2,43333	0.9	0+142232406-03	2.167437160 03
1.5	0.62500	1.70541	7.51.01	0.7	2-125141451 42	0.945671862 02
17	2.01125	1.23000	a. 49700	1.00000	0.133923266 00	-3,579194567 00
14	0.14623	1.25000	3.41100	1.00001	0.510500376-00	0.3000000446 03
1.5	0.74175	4.25930	a, \$110gg	1,49300	-0.140710417 00	0.111167111 01
20	1,44629	1.21)00	0.501111	1.00403	-0.574447906-01	-0.300173ABF 01
21	1,531/5	0.45030	0.90103	1.1000	0.25550051F 01	-0.451154146 01
"	1.65675	1,24500	9.50025	1.00000	0.4+3939177 01	-0.562425686 01
21	0,78173	4,25010	0.30300	1.00 101	7-144794457 01	10-411159444
7.	0,49763	4.12500	0.91991	1.77773	-0.9/4927545 00	-0.14474747 0/
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24	0.11175	2,25308	Lanuara	0.47309	0-104558456-01	0.110927991-01
27	0.15475	1,25300	1.01100	0.50000	0.14150534E 10	-0.291690416 00
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	1	1.1000	0.0	0.1000	11.020717	74.141011	-1	1.467739	-21.491312
	,	2.5000	4.0	9,5999	10.319186	77.614761	-1	7.604164	-14,574979
	1	1.5000	7.0	0.7309	27.449+19	20.744045			-19.167904
	•	*. >000	0.0	0.4001	11.999014	21.635734	•	7.158457	-11.430099
	4	1.5000	3.0	0.3200	67.118027	117,110750		5.477934	-34,599060
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	•	1.0000	9.5019	0,2000	7.248874	7,191329		4.531127	-4,458472
	•	1.0000	-0,5000	9.2344	-7,248794	-7,191346		4.538107	>. +>#501
	12	0.5000	-1.0000	0.1400	-1.524567	3.715967		1.249072	-2.718604
			W1)*	1 199 3					
81104	'#1.	•	ŧ	VIS' LIFT CORF		FFICIFNT HOMENT COFFECSING			IC I H N I
	ı	0.0	0.0	0.0	0.501440	1.763999		0.749470	-0.897969
	C/ = CM = CL(#OLL)	45,901 -175,101		1.470111 1.000531 0.3	C4 +	9.1100124 9.1	-n.00001# 0.0		
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4	9.36111	4.4/110	1.10000	0.0	0.419471311 01	0.276441341 02
ł	4.41444	1,14501	1.10000	U. U	-0.151016418-0/	0.314+60+56 03
•	4. /1444	4.11230	1.90303	9.0	-0.265116241 02	0.245337461 02
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•	4.91900	1.60744	2.50000	y.u	-0.465383461 01	P. 27/441811 07
•	0.15009	4,41/34	1.1090	J. J	-3.219/15991 02	0.203153041 02
,	9,04111	7, 19761	1.5-1003	9.3	U. 100778/06 0/	U. 20100763F 42
•	0.01000	5. +/416	1.32303	0.9	0.177457461 81	0.20//04441.07
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41	0.91559	A. 27943	4.41000	-1,0	0.151230041 00	0.18/9171at 02
17	0.15193	4.44753	*.*1333	9.0	-0.677070050 01	0.151-M20/4 02
1)	1.12100	7.71918	1.10000	0.0	0.741515141 05	0.75863510E 02
1.	1.4/510	1.01106	1.53901	0.3	-0.216582517 00	0. 101091741 07
13	9,12539	1,47313	7, 13003	J. J	0.32475581E 02	0. faces ? fee 0?
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11	9,13525	1,25033	9.53000	1.00000	-0.24146494 00	0,59799484 70
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22	1,05025	F. 2 1030	9.53393	1.0000	2.497137491 01	0.21952496 01
 /1	0.74175	0.27997	4.90*34	1.30303	0.161+6034+ 01	0.254579731 01
7.	3,09163	4.12500	0.50390	1.03000	0.202552998 01	-0.1/6074774 01
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7.4	1,24125	4.25000	4	0.53393	-0.24859808F 01	0.6/8497171 01
	3.60:33	5.25100	1.33490		-0.568719991 01	3.482+3804£ Q1
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*1	3,65525	1.25130	1,39340	3.53333	-11, FEATORFEE 50	0.141441478 G1
47	9,74125	4,25900	4.0000	5.41393	3.11405096F 01	
**	1, 9474.1	9,12500	1.03000	0.39800	0.107474541 07	***************************************
1*	0.99111	9.62500	1,03033	1,19911	1,/25/9/99/ 01	0.128177491 02
15	2,21125	3.25030	1.3,393	-1,53161	- J. PAD29501F-02	-0.400145161-01
17	0.15525	1.25003	E - 15 349	+0.53043	-0.154140094 00	-0.156197376 60
17	0.24125	6.25 000	1	•0•11003	-3,109456946.01	-0.674494161 01
14	3.50629	5,75509	1.01722	-0.51131		
, 1	2.511.25	4./5100	1.01137	-0.50011	0.464144516 01	-0.44/44/481 01
•)	0.45471	1,73000	1.03331	•0,55003	2. 24/224711 20	-0.141497091 01
•1	1,74177	4,25000	1.03771	-1.50377	-0.11+044141 01	-0.40/99NIR/ 00
• 7	4,49753	9.12531	1.03)50	-4.10191	-1.107914 OFF 07	-0.140470377 02
• •	0. 95111	4,52510	1,011)	-7,51103	-3,275745196 01	-0.12617727F 02
**	3,31125	7.25021	1.4)171	-1.10001		0.177744011 00
+5	7,15475	1.25000	0.41599	-1.11001	0.645365846 30	-0.397999991 00
**	0,24125	4.75000	0.1000)	-1.00001	0.754171978 00	0.811224RAF 00
• 1	9,40625	1,71000	3, 53033	-1230)	-2,159776237 11	0,201763486 01
••	9,33123	4.25900	3,59343	-1.07777		-0.510322754 00
• •	9.65525	1,75000	9.5000	+1,00043		-0.214520476 01
10	7,74125	4,24309	0.50363	• 1 . 17973		-0.259568746 81
*1	1.44143	1.17500	0.53303	-1.30000		0.124071746 01
47	0.25117	9.42400	4.43439	-1, 13001	-0.493629196-01	0.217371566 01

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	4, -110	9.1	9.5327	-4.600985	26.791831		-10.187610
•	1.5400	0. 1	1.7300	1.515256		>.66/4/6	-11-405404
•	*.595g	4.0	0.4304	2.624140	24.006371	5. #21104	-10.604745
,	1.5100				18.074368	1.5/57#9	-6.990924
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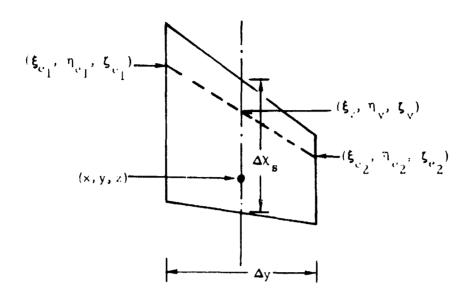
2.3 Blank and Labeled Common Blocks

2,3.1 Blank Common Block

Item No.	Mnemonics	Dimension	Symbol	Source	Description/Equation
1	NCNSM1, NBØX*			PART 1	Total number of boxes (points) on all lifting surfaces (panels) $\sum_{i=1}^{N} (nc_i^{-1}) (ns_i^{-1}),$ where N = NØPAN =
					number of panels
2	NB		,	MAIN	Total number of bodies - input
3	NDELT		δ	MAIN	Symmetry flag - input
4	NDATA, N3			MAIN	Data flag - input
5	NØPAN, NP			MAIN	Total number of panels - input
6	IQ		r	initialized in MAIN.	Index of receiving point
7	IR		s	used in PART 2	Index of sending point
8	JSPECS, JSP		€	MAIN	Second symmetry flag - input
9	FMACH, M		М	MAIN	Mach Number - input
10	NCARAY	20		PART 1	Array of the number of the chordwise divisions for all panels

^{*} A clash through the O indicates the letter O rather than the number zero.

Item No.	Mnemonics	Dimension	Symbol	Source	Description/Equation
11	NSARAY	20		PART 1	Array of the number of the spanwise divisions for all panels
12	NBARAY	20	nba _i	PART 1	$nba_{i} = \sum_{j=1}^{i} (nc_{j}-1) (ns_{j}-1),$
					where i = 1, NØPAN
13	ACAP, REFARA		A, S	MAIN	Reference area - input
14	B2, S		b/2, s	MAIN	Reference semispan-input
15	FL, REFCHD		ē, 2b _r	MAIN	Reference chord - input
16	BETA		β	MAIN	$\beta = \sqrt{1 - M^2}$, where $M = Mach Number$
17	ΡΙ		π	MAIN	π = 3.1415926, built-in constant
18	KR, RFREQ		k _r	MAIN	Reduced frequency - input as RFREQ(20)
19	KRDBR		k _r /b _r	MAIN	Normalized k
20	GMA	50	Yį	PART 1	Array of dihedral angles for all panels in radians; i = 1, NØPAN
21	х	400	x	PART 1	Array of x-coordinates of receiving points (3/4 chord point of box) - see sketch on next page.



Sketch 2.3-1

Item No.	Mnemonics	Dimension	Symbol	Source	Description/Equation
22	Y	400	у	PART 1	Array of y-coordinates of receiving points
23	ZZ	400	z	PART 1	Array of z-coordinates of receiving points
24	Zl	400	ξ _{e1}	PART 1	Array of x-coordinates
25	Ρl	400	$\eta_{f e1}$	PART 1	Array of inboard sending points
26	ZZl	400	s_{e1}	PART 1	Array of z-coordinates
27	22	400	ξ _e 2	PART 1	Array of x-coordinates
28	P2	400	η _e 2	PART 1	Array of y-coordinates points
29	ZZZ	400	ζ _e 2	PART 1	Array of z-coordinates

Item No.	Mnemonics	Dimension	Symbol	Source	Description/Equation
30	EV	400	ξ _v	PART 1	Array of x-coordinates
31	PV	400	η _v	PART 1	Array of center- y-coordinates line
32	ZV	400	ζ _v	PART 1	Array of z-coordinates
33	SDELX	400	Δx_{s}	PART 1	Array of average chordlengths of all boxes
34	DELY	400	Δy	PART 1	Array of box widths.
					See sketch above for items no. 21-34.

The Blank Common is used by the following subroutines of the pregram. MAIN, PARTI, GENQ, AUGW and PART2.

2. 3. 2 Labeled Common Blocks

Common Name	Source	User Subroutines
/ BØ DY/	PARTI	GENQ, AUGW, GENF
/MØDE/	GENQ	AUGW
/DLM/	KERNEL	INC R Ø
/NTPS/	MAIN	GENB, GENW, GENQ, PART2, SSLECX, CXSS, AIC, AERØ, FINAL, GENF
/PIGW/	PARTI	GENB
/xxz/	PARTI	FINAL, GENF
/XYZ./	PARTI	GENQ, AERØ
/YZY/	GENB	AERØ. FINAL

The following is a detailed description of the contents of each labeled common block.

Common Name	Mnemonics	Dimension	Symbol	Source	Description/Equation
	R0	100	R _{oj}	PARTI	Body radius distribution
ВФДҮ	ROP	100	R'oj	PART1	x-derivative of body radius distribution
	NBEA	20	nbe i	PARTI	$\begin{array}{c} i\\ nbe_i = \sum_{j=1}^i (nf_j-1),\\ \\ where nf_j \ is \ the \ number\\ \\ of \ body \ endpoints \ for \ the\\ \\ j-th \ body; \ i=1, \ NB \end{array}$

Common Name	Mnemonics	Symbol	Description	
	N8(80)	N8	Flag for spanwise v	rariables
	NARQ(80)	NA.	Number of ARQ val panels and modes	ues for a ll
M¢ DE	LARQ(120, 2)	(N, M)	Array of subscripts ARQ's	of all
	ARQ(120)	ARQ	Array of ARQ's	
	H(400)	[h]	[h] matrix	
	DH1(400)	[dh/dx]	[dh/dx] matrix	input for
	DH2(400)	[d ² h/dx ²]	$[d^2h/dx^2]$ matrix	special cases only
			See Input Data, Sec	2.1.2

Common Name	Mnemonics	Symbol	Source	Description/E	quation	
	K10	K ₁ (s)		Planar part of	the steady contribution to the Kernel	
	K20	K ₂ ^(s)		Nonplanar part of	to the Kerner	
	KIRTI	R ℓ K 1		R ! (ΔΕ ₁) Τ ₁)	
DLM	KHTI	$Im\overline{K}_1$	KERNEL	Im (ΔK ₁) T ₁	Unsteady part of	
	K2RT2P	R I E ₂		R I (ΔK ₂) T [*] ₂	Kernel	
	K2IT2P	$^{ m Im}\overline{E}_2$		Im (ΔK ₂) T ₂ *	·)	
	K10 T1			K ₁ (s) T ₁		
	K20T2P			K ₂ ^(s) T ₂ *		
	E2	.2	INC RØ	e semi-widt sending box	h of	

Common Name	Mnemonics	Dimension	Symbol	Source	Description/Equation
NTPS	NTPI NTP2 NTP3 NTP4 NTP5 NTP6 NTP7 NTP8 NTP9 NTP10			MAIN	Variable names for logical tape units used in program. For present assignment of units see Sec. 2.4 (Logical Tape Units)
	CTI	20	$\widetilde{\mathrm{C}}_{1}$		Array of chord- lengths for inboard edge of all panels; $\widetilde{C}_1 = X_2 - X_1$
PIGW	CT2	20	Ĉ	PARTI	Array of chord- lengths for out- board edge of all panels; C _o X ₄ - X ₃
	TS	50	āvg		Tavg = 1

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Common Name	Mnemonics	Dimension	Symbol	Source	Description/Equation
xxz	xø c	400	x/c		x-location of load points as a fraction of local chord
XXZ	ХIJ	50	x le	PART1	x-coordinate of leading edge for center of strips
	YS	50			y-coordinate of centerline of strip
	DELYS	50			Δy for strip
	ZS	50			z-coordinate of center- line of strip
XYZ	DELZS	50		PART1	Δz for strip
	FCIAMMA	50			Dihedral angle for strip, in radians
	CWIG	50			Average chord length of strip
	X1A*	50			(Alternative #1 AIC) Elastic axis (ALT, #2)
					Location of first control point.
	X3A*	50			Location of third control point.
YXY	X 5 A *	50		GENB	Location of rotation axis for control surface
IAT	X7A*	5)			Location of rotation axis for tab
	X2A*	50			Location of second control point

^{*} All coordinates given in fractions of local chord

2.4 LOGICAL TAPE UNITS

Tape	Number	
Symbol	Present Assignment	User Subroutines
NTPI	1	MAIN, GENQ, PART2, SSLECX, SØLVIT, QUAS, FUTSØL
NTP2	i 3	MAIN, QUAS, FUTSØL
NTP3	3	MAIN, GENW, GENQ, SSLECX, CXSS, SØLVIT, QUAS, FUTSØL, AIC, FINAL, GENF
NTP4 NTP7 NTP8	4 12 8	MAIN, GENQ, CXSS, SØLVIT, QUAS, FUTSØL AUGW, GENF MAIN, GENQ, QUAS, FUTSØL
NTP9	9	MAIN, GENQ, SØLVIT, QUAS, FUTSØL, AIC, AERØ
NTP10	10	MAIN, GENB, GENW, CENQ, SSLECX, CXSS, AIC, FINAL, GENF
LTAPE	11	QUAS, FUTSØL
MTAPE	7	GENQ
NAT	NTP2	QUAS, FUTSØL
RHSTAP	4	QUAS, FUTSØL

Logical tape units 5 and 6 are used throughou the program as the standard READ and WRITE units respectively.

Logical tape units MTAPE, NTP9 and LTAPE can be specified as master tapes. MTAPE can be used as modal input tape for special cases; see 2.5.6.4 for pertinent information. NTP9 is used for saving the AIC supercolumns when this is desired — see 2.5.19.6. LTAPE contains the quasi-inverse matrix for future solutions of cases with new modes only — see 2.5.17.6.



2.5 DESCRIPTION OF SUBROUTINES

Program H7WC is operational on the JBM 7094 and 360/65, and the GE 635 computers. The main part of the program controls all the twenty-one subroutines of the program: PARTI, GENB, GENW, GENQ, AUGW, TKER, PART2, INCRØ, KERNEL, IDF1, IDF2, SNPDF, SSLECX, CXSS, SØLVIT, QUAS, FUTSØL, AIC, AERØ, FINAL and GENF. The following is a description of each of the subroutines in the order of their use within the program.

2. 5. 1 MAIN

Functional Description

The MAIN part of the program reads the header card for each data set, the Mach number, the reference area, the reference semi-span and the reference chord as well as the control flags that define the various types of cases. MAIN also reads the array of reduced frequencies, and in case of an AIC-type analysis, additional control data is read, as well as the gust data, if computation of the harmonic gust coefficients is also desired. In case of a modal-type analysis, the control data is read for the modes to be used. Depending on the type of the case, MAIN selects the appropriate subroutines in an overall frequency-loop, except for PART1 and GENB (latter in case of AIC's only), which are independent of the reduced frequency.

The following is a summary of all data items used in MAIN, grouped according to their source. Detailed description of each input data item is given in Sec. 2.1.2. Other data items are described in Sec. 2.3 (Contents of Common) and/or Sec. 2.4 (Logical Tape Units) as indicated below.

Data from Cards

FMACH, ACAP, FL, B2, NDELT, NP, NB, NRF, NCØRE, N1, N2, N3, N4, RFREQ(20), NSTRIP, NB1, NB2, NPR1, NPR2, NPR3, JØBNØ, JØBNØ2, NGUST, JSPECS, NPC, NSV, NBV, NYAW, GZRØ, XZRØ, VEL, WG, LIM(50, 3), NMD, NTA, N5, N6, N7, NTP1, NTPM, NMTP, NMTB.

See Sec. 2.1.2 for description.



Data from Common

NCNSM1, NCARAY(20), NSARAY(20), NBARAY(20), GMA(50), X(400), Y(400), ZZ(400), Z1(400), P1(400), ZZ1(400), ZZ(400), P2(400), ZZ2(400), EV(400), PV(400), ZV(400), SDELX(400), DELY(400) - from Blank Common.

YS(50), DELYS(50), ZS(50), DELZS(50), FGAMMA(50), CWIG(50) - from Common /XYZ/.

X1A(50), X3A(50), X5A(50), X7A(50), X2A(50) - from Common /YZY/.

 $X \phi C(400)$, CWIG(400) = from Common /XXZ/.

CT1(20), CT2(20), TS(50) - from Common /PIGW/.

R0(100), R0P(100), NBEA(20) - from Common $/B\phi$ DY/.

K10, K20, K1RT1, K1IT1, K2RT2P, K2IT2P, K10 F1, K10 T2P, E2 - from Common / DLM/.

See Sec. 2. 3 for details.

Data from Tapes

MNEMØNICS	SYMBØL	TAPE NAME	DESCRIPTION
AUGM(530)	[A ^{ww}]	NTPI	Matrix of complex downwash factors, read one row at a time. Dimension of matrix is $NB\phi X$ x $NB\phi X$.
WQ(130)	[w _q]	NTP3	Matrix of complex polynomial mode right-hand sides, read one row (of length NMD) at a time. Dimension of matrix is NBØX x NMD.
NTPØLD		LTAPE	Identification number of case for which the quasi-inverse of the matrix of downwash factors was saved on LTAPE in a previous run. For special cases only.

See Sec. 2. 4 for present assignment of logical tape units.

Data to Common

NB, NDELT, NDATA, NOPAN, IQ, IR, JSPECS, FMACH, ACAP, B2, FL, BETA, PI, KR, KRDBR - to Blank Common.

NTP1, NTP2, NTP3, NTP4, NTP5, NTP6, NTP7, NTP8, NTP9, NTP10 - to Common/NTPS/.

See Sec. 2.3 for details.

Data to Tapes

MNEMØNICS	SYMBØL	TAPE NAME	DESCRIPTION
NTPI		LTAPE	Identification number of current case, used to identify the quasi-inverse that is to be saved on LTAPE. For special cases only, when the input data N5 is set to 1.
AUGM(530)	[A ^{ww} ¦w _q]	NTP4	Augmented downwash matrix: matrix of downwash factors augmented by the matrix of poly- nomial mode right hand sides written one row (of length NBOX + NMD) at a time. Dimension of matrix is NBØX x (NBØX + NMD).

2. 5. 2 SUBROUTINE PART1 (JCUM, CØEFP, CØEFB, YIN, ZIN)

Functional Description

Subroutine PART1 reads the geometry input of the idealized air-craft, the desired number of spanwise and chordwise divisions per panel, and the fractional location of the division lines. The panels are subdivided into boxcs, and arrays of coordinates are generated for all boxes of all panels. In the case when body interference is also desired, PART1 reads the geometry data for all bodies and generates the necessary body-coordinate arrays. The computed arrays are made available for subsequent calculations through the Blank Common block placed in the

appropriate user subroutines. Additional items of computed geometry are transmitted to MAIN via the argument list of PARTI, and to several other subroutines via the Labeled Common blocks /PIGW/, /XXZ/ and /XYZ/.

Data from Cards

XCAP(4), YCAP(2), ZCAP(2), NC, NS, CØEFP(20), TH(50), TAU(50), ZSC, YSC, NF, NZ, NY, CØEFB(20), F(50), RAD(50).

See Sec. 2. 1.2 for details.

Data from Common

NB, NDELT, NDATA, NØPAN, IQ, IR, JSPECS, FMACH, ACAP, B2, FL, BETA, PI, KR, KRDBR - from Blank Common Block.

See Sec. 2.3 for details.

Data from Tapes

None

Data to Common

NCNSM1, NCARAY(20), NSARAY(20), NBARAY(20), GMA(50), X(400), Y(400), ZZ(400), Z1(400), P1(400), ZZ1(400), ZZ(400), P2(400), ZZ2(400), EV(400), PV(400), ZV(400), SDELX(400), DELY(400) - to Blank Common block.

YS(50), DELYS(50), ZS(50), DELZS(50), FGAMMA(50), CWIG(50) - to Common /XYZ/.

 $X\emptyset C(400)$, XIJ(50) - to Common /XXZ/.

CT1(20), CT2(20), TS(50) - to Common /PIGW/.

R0(100), R0P(100), NBEA(20), BGMA(20) - to Common /BØDY/.

Data to Tapes

None

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION/EQUATION
ş		
JCUM, NSTRIP		Number of strips on all panels
$C\phi EFP(20)$		Array of panel-coefficients — input
$C\phi EFB(20)$		Array of body-coefficients — input
YIN(20)		Array of y-coordinates of inboard edge of panels
ZIN(20)		Array of z-coordinates of inboard edge of panels

2.5.3 SUBROUTINE GENB (EV. SDELX, DELY, X, NSTRIP, BETA, IMX, NPC, NP, NSARAY, LIM)

Functional Description

Subroutine GENB reads the AIC-type mode flags for all panels of the lifting surfaces. It also reads the array of fractional chord lengths for the various control points and hinge line locations at the inboard edge and at the outboard edge of each panel. Subroutine GENB computes arrays of coordinates to be used in GENB and GENW and saves them on logical tape unit NTP10. Depending on the setting of the flag NPC (input data), subroutine GENB computes the integration matrix [B] according to Alternative 1 (NPC = 0) or Alternative 2 (NPC = 1). (See Eq. 2.5-12, Pt I, Vol I.) The [B] matrix is saved on logical tape unit NTP10 in a partitioned format, and is used in subroutine AIC for the computation of the AIC supercolumns * and harmonic gust coefficient columns, when latter is also desired.

Data from Cards

 $N\phi P(20)$, IS(20, 7), XHI(20, 7), $XH\phi(20, 7)$.

^{*} A supercolumn of a matrix is a matrix-column whose elements are the columns of the original matrix taken in order; i.e., the second column is below the first, the third column is below the second, etc. A square matrix of order N becomes a supercolumn matrix of size N² x 1.

See Sec. 2.1.2 for details.

Data from Common

NTP1, NTP2, NTP3, NTP4, NTP5, NTP6, NTP7, NTP8, NTP9, NTP10 - from Common /NTPS/.

CT1(20), CT2(20), TS(50) - from Common /PlGW/.

Data from Tapes

None

Data to Common

X1A(50), X3A(50), X5A(50), X7A(50), X2A(50) - to Common / YZY/.

Data to Tapes

MNEMONICS	SYMBOL	TAPE NAME	DESCRIPTION
NSTRIP			Total number of strips on all lifting surfaces (panels)— input data.
NSIZB(j,1)			Number of AIC-modes for strip j
NSIZB(j, 2)			Number of boxes in strip j
NSIZ B(50, 2)			,
LIM(50, 3)		NEDIO	Aerodynamic box limits — input
XISLCT(50, 10)		NTP10	AIC-type modal flags — input
XH(50,7)			Array of AIC-type reference coordinates
X(400)	x)
SDELX(400)	Δx		See Blank Common
B(30,10)	[8]		Integration matrix for the Aerodynamic Influence Coefficients (AIC's).

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
EV(400)	$\eta_{f v}$	
SDELX(400)	Δ× _s	See Blank Common
DELY(400)	Δу	,
X(400)	x	,
NSTRIP		Total number of strips — input
BETA	β	See Blank Common
IMX		Maximum number of rigid body modes for case
NPC		Camber flag — input
NP, NØPAN		Number of panels — input
NSARAY(20)		Array of the number of spanwise divisions (NS) for all panels
LIM(50, 3)		Aerodynamic box limits for all strips — input

2.5.4 <u>SUBROUTINE GENW (KRDBR, NCSFTØ, NPC, NGUST, WG, GZRØ, XZRØ)</u>

Functional Description

Subroutine GENW computes the AIC-type substantial derivative matrix, [W] described in Sec. 2.5.1*. Depending on the setting of the flag NPC (input data), the [W] matrix is computed according to Alternative 1 (NPC = 0) or Alternative 2 (NPC = 1). The matrix is saved for subsequent calculations on logical tape unit NTP3 in a partitioned format, described in Sec. 2.5.2*. In the case when computation of the harmonic gust coefficients is also desired, subroutine GENW also computes the gust downwash vector, and saves it on logical tape unit NTP3.

^{*}See Part I, Vol I

Subroutines GENB and GENW are used only if an AIC-type analysis is desired.

Data from Cards

None

Data from Common

YS(50), DELYS(50), ZS(50), DELZS(50), FGAMMA(50), CWIG(50) - from Common /XYZ/.

Contents of Common /NTPS/.

Data from Tapes

NSTRIP, NSIZB(50, 2), LIM(50, 3), XISLCT(50, 10), XH(50, 7), X(400), SDELX(400) - from NTP10, written in GENB.

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE NAME	DESCRIPTION
GRHS(400)	W gust	NTP3	Gust right hand sides (downwash vector) W = cos(rg-Yi) + kr (xi-xo)), where r = gust field declination from the vertical — input Y = dihedral angle of lifting surface at point i kr = reduced frequency — input br = reference semichord xi = x-coordinate of downwash point i xo = gust reference coordinate —

MNEMONICS	SYMBOL	TAPE NAME	DESCRIPTION
W(30, 10)	[W _{AIC}]	NTP3	Partitioned AIC-type downwash matrix. See Sec. 2.5.2*.

Data in Argument List

MNEMØNICS	SYMBØL	DESCRIPTION	
KRDBR		k _r /b _r	
NCSFTØ, NBØX		Total number of boxes on a panels	11
NPC		Camber flag	,
NGUST		Gust flag)
wg	Wg	Harmonic gust amplitude	input
GZRØ	$\Gamma_{\!$	Gust angle of declination	
XZRØ	×o	Gust reference coordinate	/

2.5.5 SUBROUTINE GENQ (N4, N5, N6, NRF, JRF, NBE, NMD, NMTP, NMTB, NCORE, KDO2, YIN, ZIN, COEFP, COEFB)

Functional Description

Subroutine GENQ generates the normalwash boundary condition for all lifting surface elements. It also generates the integration matrix used to obtain the generalized forces. The normalwash \mathbf{w}_{ri} (r ranges over all lifting surface box receiving points while i ranges over all modes) is written on tape row by row ($\mathbf{r} = 1$, i = 1, 2, ..., NMD, $\mathbf{r} = 2$, i = 1, 2, ..., NMD, etc.) to facilitate its use with subroutine SØLVIT. However, if the number of modes times the number of unknowns (NCØRE) exceeds the size of the working array in SØLVIT, and/or if the input flag N5 is set to 1, the W matrix is written on tape in column order, preceded by the number of modes (NMD), ready for use by subroutines QUAS and FUTSØL. The integration matrix \mathbf{B}_{ri} is written on tape in column order.

The formula for w used is given by equations $2.3-2^*$ and $2.3-3^*$ for cases without bodies, and by equation $2.2-13^*$ for body cases; the contribution of the body effect to the normalwash, w_{B_r} , is computed in subroutine AUGW, which is called from subroutine GENQ for body cases only. w_{B_r} is computed one column at a time, and is transmitted to GENQ via the argument list of subroutine AUGW. The total normalwash matrix for body cases, $(w_r - w_{B_r})_i$, is then written on tape in either row, or column order depending on the size $(NC \not QRE)$ of the problem and/or on the setting of the input flag N5.

Data from Cards

NR, NQ, NARQ(80), N8(80), LARQ(120, 2) and ARQ(120) for standard polynomial cases (N6 = 0).

When modes are input from cards (N6 \approx 2), the data input from cards are:

NBOX, BQ(400), H(400) and SH1(400).

*See Part I, Vol I

Data from Common

Contents of Blank Common and the labeled common blocks /NTPS/ and /BODY/.

Data from Tapes

N6 = 0 cases;

None, if NRF = 1. If NRF > 1 and JRF > 1 (see Sec. 2.5.5.), the following data items are read from tape NTP10: BQ(400), N8(80), NARQ(80), LARQ(120,2), and ARQ(120).

N6 = 1 cases: modes are input from tape as follows.

NBOX, BQ(400), H(400) and DH1(400) from tape MTAPE; present assignment is NTP8 = 8.

Data to Common

N8(80), PARQ(80), LARQ(120, 2), and ARQ(120), H(400), H1(400) and DH2(400) to common /MODE/.

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
BQ(400)	[B _q]	NTP10	Integration matrix for modal analysis written in column order. Dimension of matrix is NTØTxNMD, where NTØT is the total number of elements (fig. NTØT = NBØX, if there are no body elements, and NTØT = NBØX + NBE otherwise where NBE = total number of body elements).
NMD	NM	NTP4	Number of modes; written only ([N5]: Land/or NCØRE ≥ 4000.

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
WQ(400)	[W _q]	NTPW	Matrix of polynomial mode right hand sides (normalwash). Dimension of matrix is NBØX x NMD. The matrix [W] is written on tape in row order, unless N5 = 1, and/or NCØRE ≥ 4000; in latter case, [W] is written on tape in column order. Note that NTPW = NTP3 in case of row order, and NTPW = NTP4 in case of column order.
BQ(400)	[B _q]		The following items are written on tape only if more than one reduced frequency is prescribed for the case (NRF>1).
N8(80)		NTPB	See Input Data, 2.1.2
NARQ(80)	NA		
LARQ(120, 2)	(N, M)		Note that NTPB = NTP2 for cases without bodies, and NTPB = 10
ARQ(120)	ARQ		for body cases.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
N4 N5		Print flag - input Flag to save quasi-inverse - input
N6	}	Modal input data flag - input
NRF		Number of reduced frequencies for case - input
JRF		Sequence number of reduced frequency currently used in the frequency do-loop

MNEMONICS	SYMBOL	DESCRIPTION
NBE		Total number of body elements
NMD	NM	Number of modes - input
NMTP	МР	Total number of ARQ - values for all panels and modes - input
NMTB	МВ	Total number of ARQ - values for all bodies and modes - input
NCØRE		Total number of unknowns for case - input
KDØ2	KD/2	4000 - built in constant (MAIN)
YIN(20)	Y1	Array of y-cool inates of inboard edge of panels
ZIN(20)	Z1	Array of z-coordinates of inboard edge of panels.
CØEFP(20)	coef	Array of modal coefficients for panels
CØEFB(20))	Array of modal coefficients Eq.(2.3-3) for bodies

2.5.6 SUBROUTINE AUGW (NMD, NBOX, NBE, NMTB, N4, N6, NRF, JRF, J, COEFB, ASUM)

Functional Description

This subroutine gives the incremental normalwash, w_{B_r} , at all lifting surface boxes, r, due to the axial singularities on all bodies, B. Slender body theory is used to obtain the strength of the axial singularities. The formulae used in this subroutine are given by Equations (2.2-8) through (2.2-10). The incremental normalwash values, w_{B_r} , are transmitted to subroutine GENQ via the argument list under the mnemonic name ASUM one column. (for one mode) at a time. This subroutine also computes the interration matrix of all body elements for the calculation of the generalized for a for body cases.

Subroutine AUGW is bypassed for data without body elements.

Data from Cards

None for the special case when N6 = 1 (data input from tape MTAPE).

For standard polynomial cases (N6 = 0) data input from cards is same as for subroutine GENQ, except that data is now read for body elements instead of panel boxes.

Data from Common

Contents of Blank Common, and the labeled common blocks $/B\phi DY/$ and $/M\phi DE/$.

Data from Tapes

None for standard polynomial mode cases. For the special case when the modal data flag N6 = 1, the following data items are input from tape:

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
NBE			Number of body elements
BB(80)	[B _q]	NTP8	Integration matrix for the body elements
H(400)	[h]		[h] matrıx
DH1 (400)	[dh/dx]		[dh/dx] metrix
DH2(400)	$[d^2h/dx^2]$		[d ² h /ðx ²] matrix
			The above three matrices are read in column order. Dimension of matrices is NBE x NMD.

Data to Common

None

Data to Tapes

MNEMØNICS	SYMBØL	TAPE	DESCRIPTION
BB(80)	Bq	NTP8	One column of the integration matrix for body elements written on tape for one mode; length of column is NBE.
			The following items are written on tape only if more than one reduced frequency is prescribed for case (NRF > 1)
N8(80)	N8		
	ļ	NTPB	See Input Data 2.1.2
NARQ(80)	NA		
LARQ(120, 2)	(N, M)		
ARQ(120, 2)	ARQ		

Data in Argument List

MNEMØNICS	SYMBØL	DESCR1PTIØN
NMD	NM	Number of modes input
ивфх		Total number of boxes on all lifting surfaces
NBE		Total number of body elements
NMTB	МВ	Total number of ARQ values for all bodies and modes - input

MNEMONICS	SYMBOL	DESCRIPTION	
N4		Print flag for writing the [APZ], [APY] matrices	
N6		Modal input flag inp	ut
NRF		Number of reduced frequencies for case	
JRF		Sequence number of current frequency in the frequency do-loop	
J		Sequence number of current mode in modal do-loop	
C Ø EFB(20)	coef	Array of scale factors for body deflection modes.	
ASUM(400)		Incremental [W] due to body interference	

2.5.7 <u>SUBRØUTINE TKER (X0, Y0, Z9, KR, BR, GAMS, GAMSIG, KKR, KKI, FMACH)</u>

Functional Description

Subroutine TKER evaluates the total unsteady kernel for one receiving-point/sending-point combination for body cases only. (See Eq. A. 5.)* Subroutine TKER is essentially the same as subroutine KERNEL, except that the total, rather than the incremental oscillatory kernel is computed here, and that the planar and nonplanar parts of the kernel are not separated. Subroutine TKER is called from subroutine AUGW; it is bypassed in the case when the input data contains no bodies.

Subroutine TKER uses no card-, tape-, or common-data input or output; data transmission to and from TKER is done through its argument list.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
Х0	x _o	× _i - ξ _i
Υ0	Уо	$y_i - \eta_i$ See Blank Common
Z 0	z _o	$z_i - \xi_i$
KR	k _r	Reduced frequency
BR	b _r	Reference semi-chord
GAMS	Ys	Dihedral angle of sending point j
GAMSIG	Yr	Dihedral angle of receiving point i
FMACH, M	M	Mach Number
E2		e ² , where e is semi-width of sending box
KKR		Real part of total (steady + unsteady) kernel
KKI		Imaginary part of total kernel

2.5.8 <u>SUBROUTINE PART2 (NYAW, NBV)</u>

Functional Description

Subroutine PART2 is used for the calculation of the matrix of downwash factors, [D]. (See Eq. 2.1-14.*) It prepares the arguments to subroutines SNPDF and INCRØ, which do the actual computation of the downwash factor elements, and calls these in a do-loop for all boxes of all lifting surfaces. The steady part of the downwash factor elements is computed in subroutine SNPDF, and the incremental oscillatory part in subroutine INCRØ; the call to INCRØ is bypassed for steady cases ($k_r = 0.0$). Subroutine PART 2 also accounts for the effect of the (two) symmetry plane(s).

Data from Cards

None

Data from Common

Contents of Blank Common, Common /NTPS/, and Common /DLM/.

Data from Tapes

None

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
AWW(400) AWWI(400)	[A ^{ww}],	NTPl	Matrix of downwash factors written on tape in row order. Dimension of matrix is NBØX x NBØX. The real parts of the matrix elements are called AWW in the program, and the imaginary parts are called AWWI.

Data in Argument List

yaw flag NYAW

number of boxes on vertical panels NBV (fin) of T-tail case:

input in MAIN

SUBROUTINE INCRO (AX, AY, AZ, AX1, AY1, AZ1, AX2, AY2, 2.5.9 AZ2, GAMS, GAMSIG, LHS, IR, IØ, NBXS, NCPNB, NDBLE, NBV, DELR, DELT, FL,

BETA, SDELX, DELY, KR)

Functional Description

Subroutine INCRØ prepares the argument for the subroutines KERNEL, IDF1 and IDF2. It calls subroutine KERNEL to compute the incremental oscillatory part of the kernel, K, for each receiving-sending box combination at the three points of the bound vortex segment: at the center (K_C) , at the inboard point (K_i) and at the outboard point (K_O) . Since even a relatively small case requires many kernel computations (e.g., a wing with 100 boxes, in symmetry, requires $100 \times 300 \times 2 = 60,000$ kernel values), extra programming effort was made to save on computing time. Neighboring strips within a wing panel have a common kernel on the common boundary; this fact is utilized in INCRØ so that the computing time of the kernels is reduced by about 30 percent.

After the triplet of kernels, (K_c, K_i, K_o) , is obtained for one receiving-sending box combination, subroutine INCR ϕ computes the coefficients of the parabolas for the numerical integrations done in two parts. The planar part of the integration is done in subroutine IDF1, and the nonplanar part in IDF2.

Subroutine INCRO uses no card-, or tape input or output; transmission of data is done through its argument list and the labeled common /DLM/.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION/E	QUATION
AX	× _o	x - \$ _v \	
AY	y _o	y - η _v	
AZ	z _o	z - Ç _v	
AX1		x - § _{el}	
AYI		y - η _{el}	See Sec. 2-3.1 (Blank Common)
AZ1		z - ζ _{el}	(Diana Condition)
AX2		x - ge2	
AY2		y - η _{e2}	
AZ2		$z - \zeta_{e2}$	

MNEMONICS	SYMBOL	.DESCRIPTION/EQUATION
GAMS	Y _r	Dihedral angle of receiving point i
GAMSIG	Υ _s	Dihedral angle of sending point j
LHS	-	Flag; 0 when effect of right wing is computed, 1 for left-wing
IR	j	Index of sending point
ΙØ		An index running from 1 through NCM1, where NCM1 is the number of chordwise boxes of the strip in which the sending point lies.
NBXS	·	The number of boxes on the wing panel in which the sending point lies + the total number of boxes of the preceding panels.
NCPNB		The number of boxes in the first strip of the panel in which the sending point lies + the total number of boxes of the preceding panels.
NDBLE, JSPECS, JSP	€	Second symmetry flag - input
NBV		Total number of boxes on vertical panels of T-tail cases - input
DELR		Real part of the incremental oscillatory downwash factor
DELI		Imaginary part of the incremental oscillatory downwash factor
FL, REFCHD	c	Reference chord - input
BETA	β	$\sqrt{1-M^2}$
SDELX	ΔX _s	Average box-chord of sending box
DELY		Ze, box width of sending point
KR	k _r	Reduced frequency

2.5.10 <u>SUBROUTINE KERNEL (X0, Y0, Z0, KR, BR, GAMS, GAMSIG, EPS, T1, T2, M)</u>

Functional Description

Subroutine KERNEL evaluates the modified incremental oscillatory kernel, \overline{K} , for one receiving-sending point combination at a time. The planar and nonplanar parts of the kernel are evaluated separately and are returned to the calling program, subroutine INCRØ, via the labeled common /DLM/. (Evaluates right hand side of Eq. (2.1-13)*)

Subroutine KERNEL uses no card or tape input or output; transmission of data is done through the argument list and the labeled common /DLM/.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION/EQUATION
X0	× _o	AX for center AX1 for inboard edge AX2 for cutboard edge
Y0	y _o	AY for center AY1 for inboard edge AY2 for outboard edge 2.5.9
. Z0	z 0	AZ for center AZ1 for inboard edge AZ2 for outboard edge
KR	k _r	Reduced frequency
BR	b _r . c/2	Reference semichord
GAMS GAMSIG	Yr Ys	sec 2.5.9

MNEMONICS	SYMBOL	DESCRIPTION/EQUATION
EPS	€	ϵ = 0.00001; built in constant from INCR ϕ
TI	T	$T_1 = \cos(\gamma_r - \gamma_s)$
T2	T ₂	$T_2 = [z_0 \cos \gamma_r - y_0 \sin \gamma_r)(z_0 \cos \gamma_s - y_0 \sin \gamma_s)]/r^2$
M, FMACH	М	Mach number

2.5.11 SUBROUTINE IDF1(EE, E2, AT2, ETA01, ZET01, ARE, AIM, BRE, BIM, CRE, CIM, R1SQX, XIIJR, XIIJI)

Functional Description

Subroutine IDF! performs the integration of the planar parts of the incremental oscillatory kernels; see Eq. B. 9.* The result of the integration is the complex number (XIIJR, XIIJI), returned to subroutine INCRØ via the argument list of IDF1.

Subroutine IDF1 uses no card-, tape- or common input or output; data transmission to and from IDF1 is done through its argument list.

Data in Argument List

SYMBOL	DESCRIPTION
e, Δy _s /2	Semiwidth of sending box j
e ²	
T2	See 2. 5. 10
$\overline{\mathbf{y}}$	$y_0 \cos \gamma_r + z_0 \sin \gamma_r$
\overline{z}	z _o cos Y _r - y _o sin Y _r
	e ² T2 <u>ÿ</u>

MNEMONICS	SYMBOL	DESCRIPTION
ARE AIM BRE BIM CRE CIM	Re(A ₁) Im(A ₁) Re(B ₁) Im(B ₁) Re(C ₁)	Coefficients of the parabola for planar part of kernel-integration
RISQX	r 2	$\frac{\overline{y}^2 + \overline{z}^2}{2}$
XIIJR		Real part of integral planar
хшл		Imaginary part of integral contribution

2. 5. 12 SUBROUTINE IDF2 (EE, E2, AT2, ETA01, ZET01, A2R, A2I, B2R, B2I, C2R, C2I, R1SQX, DIIJR, DIIJI)

Functional Description

Subroutine IDF2 performs the integration of the nonplanar parts of the incremental oscillatory kernel; see Eqs. (B. 15) and (B. 16).*

The result of the integration is the complex number (DIIJR, DIIJJ), returned to subroutine INCRØ through the argument list of IDF2.

Subroutine IDF2 uses no card, tape or common input or output; data transmission to and from IDF2 is done through its argument list.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
EE	e	
E2 '	e ²	
AT2	T ₂	See 2.5.11
ETA01.	<u>y</u>	
ZET01	z	j
A2R	Re(A ₂)	
A2I	Im(A ₂)	Coefficients of the parabola
B2R	R e(B ₂)	for nonplanar part of kernel integration
B2I	Im(B ₂)	
C2R	Re(C ₂)	
C2I	Im(C ₂))
RISQX		See 2.5.11
DIIJR		Real part of integral nonplanar
DINI		Imaginary part (contribution of integral

2.5.13 SUBRØUTINE SNPDF (SL,CL,TL,SGS,CGS,SGR,CGR, X0,Y0,Z0,EE,DIJ,BETA,CV)

Functional Description

Subroutine SNPDF computes the steady downwash factors for one receiving-sending box combination at a time. The result, DIJ, is returned to PART2 via its argument list. (See Eq. C. 35b, Part I, Vol I.)

Subroutine SNPDF uses no card, tape, or common data input or output; data transmission to and from SNPDF is done through its argument list.

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Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
SL		sin A, where A is the sweep angle of the 1/4-chord line of sending box [sweep angle of bound vortex]
CL		modified by Prandtl-Glauertrule.
TL		tan A
sgs		sin Y _s
CGS		cosy
SGR		sin Y _r
CGR		cos Yr
X0		
YO		See 2.5.10
20		
EE	e	Semi-width of sending box
DIJ		Steady contribution to downwash factor
BETA	β	$\sqrt{1-M^2}$
CV	c _{avg} ,∆X _s	Δx, average chord length of sending box

2. 5. 14 SUBROUTINE SSLECX (NCNSM1, MD, NGUST)

Functional Description

Subroutine SSLECX is used with AIC-type data, and only if $NC\emptyset RE < 4000$. It reads the matrix of downwash factors [D] from tape NTP1 one row at a time, and augments it with the AIC-type downwash matrix [W]_AIC read from tape NTP3. For gust cases, when NGUST ± 0 , subroutine SSLECX also reads the gust downwash vector from tape NTP3. The augmented matrix [D|W] is saved on tape NTP4 in row order, ready to be used by subroutine SØLVIT to obtain the solutions to the system of simultaneous linear equations represented by the matrix [D|W].

Data from Cards

None

Data from Common

Contents of Common /NTPS/.

Data from Tapes

SYMBOL	TAPE	DESCRIPTION
	NTP10	See 2. 5. 3
[A]	NTPl	Matrix of downwash factors written on tape in row order in subroutine PART2. Dimension of matrix is NBØXxNBØX.
$\left\{\mathbf{w}_{\mathtt{gust}}\right\}$	NTP3	Gust downwash vector
[w] _{AIC}	NTP3	AIC type downwash $\begin{cases} See \\ 2.5.4 \end{cases}$ (normalwash).
	$\{W_{gust}\}$	NTP10 [A] NTP1 {Wgust} NTP3

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
AUGM(530)	[A W]	NTP4	Augmented downwash factor matrix

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
NCNSM1	n	Total number of boxes on all lifting surfaces (panels)
MD	m	Total number of AIC-type right-hand sides $NSTRIP$ $m = \sum_{j=1}^{XISLCT} XISLCT_{jmax}$
		where NSTRIP = total number of strips on all lifting surfaces, and XISLCT = the number of elastic modes for j-th strip - whenever NGUST = 0. When NGUST \pm 0 NSTRIP $m = \sum_{j=1}^{NSTRIP} XISLCT_{jmax} + 1.$
NGUST		Gust flag - input

2. 5. 15 SUBROUTINE CXSS (NCNSM1, MD, NGUST)

Functional Description

Subroutine CXSS is used with AIC-type data, and only if NCØRE ≥ 4000, and/or N5 = 1. It reads the AIC-type downwash matrix, [W]_{AIC}, from tape NTP3 and forms columns of right-hand sides. When NGUST ≠ 0, subroutine CXSS also reads the gust downwash vector from tape NTP3; this becomes the last column of right hand sides. These are saved on tape NTP4 preceded by the number of right hand sides for the case, 29 ady to be used by subroutine QUAS and FUTSØL to obtain the solutions.

Data from Cards

None

Data from Common

Contents of Common /NTPS/.

Data from Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION	
NSTRIP NSIZB(50, 2) LIM(50, 3)		NTP10	See 2.5.3	
GUST(400) W(30,10)	{w _g }	NTP3 NTP3	Gust downwash vector Partitioned matrix of AIC-type down- wash (normalwash)	see 2.5.4

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
NMD	m	NTP4	Number of AIC-type right hand sides (elastic modes), including the gust right hand side whenever NGUST ‡ 0
WX(400))	Matrix of AIC-type right hand sides written in column order. Dimensions of matrix is NBØX x NMD

Data in Argument List

Same as in 2, 5, 14

2.5.16 <u>SUBROUTINE SOLVIT (ND, MD, KD, NI, NM, NØ, NW, NPRI)</u>

Functional Description

Subroutine SØLVIT is used for cases with NCØRE < 4000, and/or N5 = 0. It solves the system of simultaneous linear equations represented by the augmented matrix [D |W|, written on tape NTP4. The solutions are saved on tape NTP3 in column order; the data on the input tape, NTP4, is not preserved.

Data from Cards

None

Data from Common

None

Data from Tapes

AUGM(530), the augmented downwash factor matrix from tape NTP4 - see 2.5.1 or 2.5.14.

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
A(4000)		NTP3	Solution matrix written in column order; dimension of matrix is NDxMD, where ND = NBØX, and MD = NMD.

Note that subroutine SØLVIT uses a single complex work array, A(4000), for all tape input and output variables.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
ND, NCNSM1, NBØX	n	Total number of boxes on all lifting surfaces (panels) = size of the square matrix $[A^{ww}] = [D] = matrix$ of downwash factors.
MD	m	Total number of right hand sides in the system of simultaneous linear equations solved in subroutine SØLVIT.
КD		Working array size (real-variable dimension); presently set to 8000
NI		NTP4; tape number assigned to tape containing all rows of the augmented matrix [A ^{ww} W] - see 2.5.1 or 2.5.14
NM		NTP1; tape number, used as scratch unit
NØ		NTP9; tape number, used as scratch unit
NW		NTP3; tape number assigned to tape containing all solutions of the system of simultaneous linear equations
NPR1		Print flag for solutions - input

For present assignment of tape units see 2.4.

2. 5. 17 SUBRØUTINE QUAS (ND, MD, KD, NI, MM, NØ, NAT, NW, LTAPE, RHSTAP, NPR1)

Functional Description

Subroutine QUAS is used in conjunction with subroutine FUTSØL for cases with NCØRE ≥ 4000, and/or N5 = 1. It reads the matrix of downwash factors, [D] from tape NTPl in row order, and as many columns of right hand sides (m,say), from tape RHSTAP, as the dimension of the work area in QUAS (3700 complex words) will permit. Subroutine QUAS computes the so called quasi-inverse of the matrix [D] and saves it on tape LTAPE; it also computes the solutions for the m right hand sides and saves these on tape NTP3.

Data from Cards

None

Data from Common

None

Data from Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
A(3700)	[D], [A ^{ww}]	NTPl	Matrix of downwash factors; see 2.5.8
NMD		NTP4	Total number of modes for case
A(3700)	[w]	NTP4	Matrix of right hand sides; see 2.5.5 or 2.5.15

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
A(3700)		LTAPE	Quasi-inverse matrix of [D]. Master tape has to be specified to preserve quasi-inverse for future analysis; otherwise tape is considered a scratch unit.
A(3700)		NT P3	Solution matrix; dimension NBØX x n, where n is the number of right hand sides for which solutions were obtained.

Note that Subroutine QUAS uses a single complex work array, A(3700) for all input and output variables.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
ND, NCNSM1, NBØX	n	Total number of boxes on all lifting surfaces
MD	m	On output MD = n = the number of right hand sides for which solutions were obtained
KD		Working array size (complex variable dimension); presently set to 3700.
NI		NTP1; tape number assigned to tape containing all rows of matrix [A ^{ww}] (matrix of downwash factors)
мм		NTP8
NØ		NTP9 tape numbers for scratch units
NAT		NTP2

MNEMONICS	SYMBOL	DESCRIPTION
NW		NTP3, tape number for the solution matrix
LTAPE		Tape number assigned to the tape containing the quasi-inverse of [D]
RHSTAP		NTP4; tape number assigned to tape containing the matrix of right hand sides

For present assignment of tape units see Sec. 2.4

2.5.18 SUBROUTINE FUTSOL (ND, MD, KD, NI, MM, NO, NAT, NW, LTAPE, RHSTAP)

Functional Description

Subroutine FUTSØL is used for cases with NCØRE \geq 4000, and whenever new modes only are required for a previously computed case (Nl = 1). When Nl = 0, subroutine FUTSØL is called following subroutine QUAS, and uses the quasi-inverse of the matrix [D] from tape LTAPE. Additional right hand sides are read from tape RHSTAP, and the additional solutions are saved on tape NTP3 following the previously obtained solutions.

Subroutine FUTS \emptyset L is bypassed when N5 = 1 and NC \emptyset RE < 3700.

Data from Cards

None

Data from Common

None

Data from Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
A(3700)	[w]	RHSTAP	Additional columns of right hand sides
A(3700)		LTAPE	The quasi-inverse matrix generated in QUAS

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
A(3700)		NW	The additional solutions for the additional right hand sides

For present assignment of tape units see Sec. 2.4.

Data in Argument List

Same as for QUAS - see 2.5.17, except that NI is now merely a scratch unit and does not contain the input matrix [D].

2. 5. 19 SUBROUTINE AIC(BR.S. KR. NBF. NBL. JØBNØ, ICØD. VEL. NGUST. NPR2)

Functional Description

Subroutine AIC is used for AIC-type cases only. It reads the solutions to the AIC-type elastic modes from tape NTP3 and the partioned integration matrix, [B], from tape NTP10. Using these arrays subroutine AIC supercolumns according to Eq. 2.5-1*, and saves them on tape NTP9.

Data from Cards

None

Data from Common

Contents of the labeled common blocks /NTPS/, /XYZ/ and /YZY/.

Data from Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
NSTRIP			ν.
NSIZE(50, 2)	}	NTP10	See 2. 5. 3
LIM(50, 3))		
B(10, 30)	[B] _{AIC}		Integration matrix
DIJM1W(400, 4)	$[D]^{-1}[W]_{AIC}$	NTP3	Solution matrix for the AIC type right hand sides
CPGUST(400)		N°f P3	Solution for the gust right hand side - read only if NGUST \neq 0.

Data to Common

None

Data to Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
CH(4, 4, 50)		NTP9	AIC matrix written one supercolumn at a time; see Eq. 2.6-2.*

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
Bk	ь _r , с/2	Reference semichord
s, REFSPN	ե _{/2} , s	Reference semispan - input
KR	₩,	Reduced frequency
NBF		First bay number for which
NBL		Last bay number AIC's are written on
Ј Ф ВИ Ф		AlC-case-identifier tape
IC ∲ D		l, built in constant
VEL		Aircraft velocity
NGUST		Gust flag - input
NPR2		Print flag for A!C's

2. 5. 20 SUBROUTINE AERO (BR. B2, KR. ACAP, NRF, NSV, NYAW, IMX, NPC, NDELT, NPR3)

Functional Description

Subroutine AERØ is used for AIC-type cases only. It reads the AIC supercolumns from tape NTP9 and computes the stability derivatives for the rigid body modes according to Eqs. (2.4-6) - (2.4-12). This subroutine is primarily used for the monitoring of the AIC's. It can be bypassed by setting the data flag NPR3 to 0.

Data from Cards

None

Data from Common

Contents of the labeled commons /NTPS/, /XYZ/ and /ZYZ/.

Data from Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
NSTRIP			
NSIZB(50, 2)			
LIM(50, 3)		NTP10	See 2. 5. 3
XISLCT(50, 7))		
CH(4, 4, 50)		NTP9	See 2.5.19

Data to Common

None

Data to Tapes

Usually none. However, in special cases, when dynamic stability derivatives are also required (input flag NPR3=3), tape unit 2 is used for saving the total lift-and moment coefficients computed for the steady case (KR(1) = 0.0). Subsequently, this information is read from tape unit 2 and is used in the calculation of the dynamic stability derivatives for the unsteady case (KR(2) > 0.0) - see 2.4.2, in Part I, Vol I.

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
BR	b _r , c/2	
B2	ь ₂ , s	See 2. 5. 19
KR	k _r	
ACAP, REFARA	A, S	Reference area - input

MNEMONICS	SYMBOL	DESCRIPTION	
NRF		Number of reduced frequencies for case - input	
NSV		Number of strips (fin) of T-tail cas - input	on vertical panels es; 0 otherwise
NYAW		Yaw flag - input	
IMX		Number of rigid body modes for AIC-type case-from GENB	
NPC		Camber flag)
NDELT	δ	Symmetry flag	- input
NPR3		Print flag	

2.5.21 SUBROUTINE FINAL (BR, B2, ACAP, NSV, NBV, IMX, NYAW, NPC, NDELT, FGAMMA)

Functional Description

Subroutine FINAL is used for AIC-type cases only. It reads the solutions for the AIC-type right hand sides from tape NTP3 and computes the pressures for the rigid body modes for all boxes of all lifting surfaces. These are printed along with the box number and the fractional chordwise location of the load points (1/4-chord of box). Subroutine FINAL can be bypassed by setting the data flag NPR3 to 0.

Data from Cards

None

Data from Common

Contents of the labeled commons /NTPS/, /XXZ/, and /YZY/.

Data from Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
NSTRIP)	
NSIZB(50, 2)		NTP10	See 2.5.3
LIM(50, 3))	
CPM(400,4)		NTP3	Solution matrix for the AIC-type right hand sides

Data to Common

None

Data to Tapes

None

Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
BR	b _r	
B2	b∕2, s	See 2.5.20
ACAP, REFARA	A, S	
NSV		1
NBV		Total number of boxes on vertical panels (fin) of T-tail cases - input
IMX)
NYAW		See 2.1.2
NPC		()
NDELT	δ	1
FGAMMA(50)		Array of dihedral angles for all strips of all lifting surfaces.

2. 5. 22 SUBROUTINE GENF (NBOX, NTOT, NMD, NDELT, NST, LIM, FL, XCAP, XLI, EV, SDELX, B2, Y, ZZ, XOC, NBV)

Functional Description

Subroutine GENF is used for modal type cases only. It reads the solutions for all modes from tape NTP3 and writes the pressures for all modes and all boxes along with the x, y, z coordinates of the boxes. It also computes the sectional lift and moment coefficients for all strips of all lifting surfaces as well as the total lift- and moment coefficients - see Eqs. (2.4-6)-(2.4-12)Subroutine GENF also computes the generalized forces for all pressure modes and deflection modes; see Eq. 2.4-5.*

Data from Cards

None

Data from Common

Contents of the labeled common blocks /NTPS/, /XYZ/ and /B ϕ DY/.

Data from Tapes

MNEMONICS	SYMBOL	TAPE	DESCRIPTION
P(400)	$\left\{\Delta C_{p}\right\}$	NTP3	Solution matrix for modal type data read in column order. Dimension of matrix is NBØXxNMD.
P(400)	$\left\{\Delta C_{p}^{(f)}\right\}$	NTP7	Matrix of pressures for all body elements and modes read in column order. Dimension of matrix is NBExNMD. This data is read for body cases only.
B(400)		NTP10	Integration matrix for modal type data read in column order. Dimension of matrix is NTØTxNMD where NTØT = NBØX + NBE.

Data to Common

None

Data to Tapes

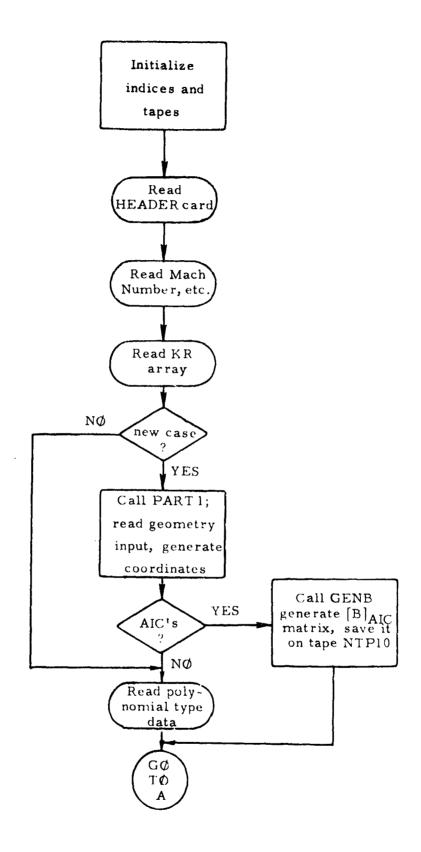
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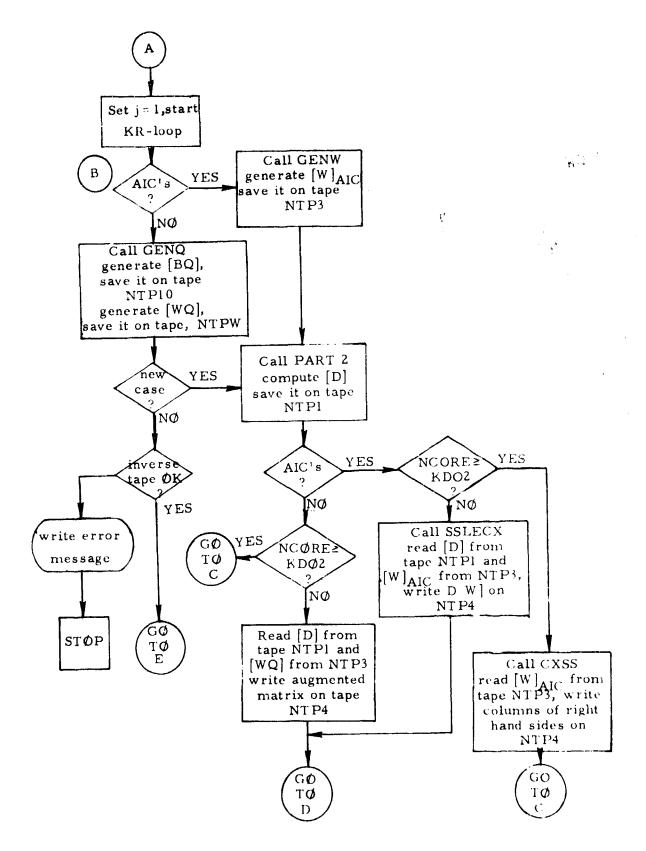
Data in Argument List

MNEMONICS	SYMBOL	DESCRIPTION
NDELT NB NSTRIP	δ	Symmetry flag Number of bodies Number of strips
ивфх		Total number of boxes on all panels
NTØT	n	NBØX + NBE, where NBE = total number of body elements
NBV		T toil input
NSV	·	T-tail input
NMD	NM, m	Number of modes - input
LIM(50, 3)		See Sec. 2.5.3
ACAP, REFARA	A, S	
FL, REFCHD	ē, 2b _r	
B2, REFSPN	s, b/2	
EV(400)	ξ	
Y(400)	у	See Blank Common, Sec. 2.3.1
ZZ(400)	z	
SDELX(400)	Δxs	
XIJ(50)	x _{le}	See Common /XXZ/, Sec. 2, 3.2
XØC(400)	\$ /c	See Common / AAZ/, Sec. 2, 3.2

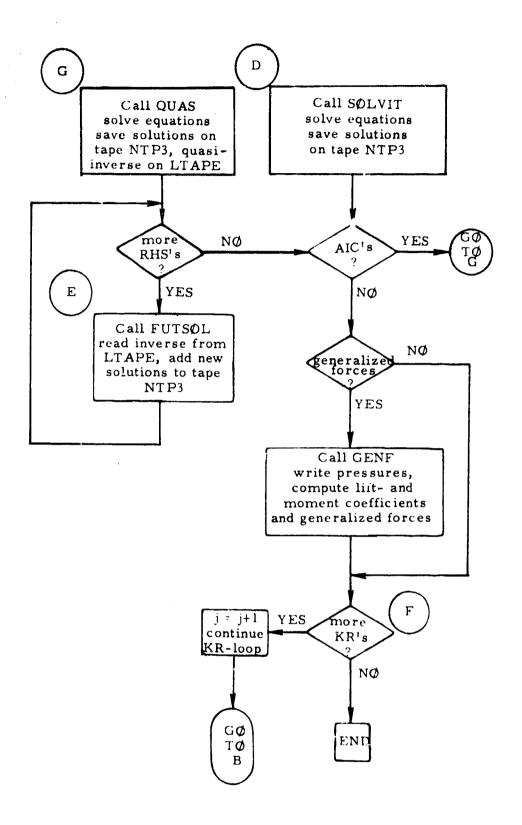
2.6 BLOCK DIAGRAM

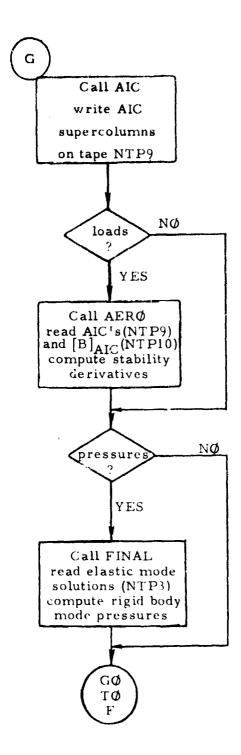
The basic block diagram for the computer progrm H7WC is shown on the next four pages. All logical steps in the "main" part of the program are indicated, and the basic functions of all subroutines that are called from "main" are outlined. A complete description of all subroutines—including those that are not called directly from "main," and therefore are not indicated in the block diagram—can be found in Sections 2.5.1 through 2.5.22.





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2.7 PROGRAM LISTING

The following pages (pp. 95 - 143) contain the complete listing of program H7WC. The individual subroutines of the program (including the "main" part of it) are listed in the order of their description in Section 2.5.

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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   FGAMALJ=GMATILOJP)
620 N + N + NCM1
M1 + NSOFAR + 1
PERTICOLO

PRINCIPADO

PRINCIP
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          INDEX -K- FOUALS SUBSCRIPT (Wel, N)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           INDEX -L- EQUALS SUBSCRIPT IN, N+1)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        THOEX -J- EQUALS SUBSCRIPT(4.N)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 4 -tH-13 - NC + (1-13
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         KE . INel: . NC . N
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##17F 16,3301 11.0uP	PRT12680	25.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.
MAITE (6,170) (CHIGH), 1-11,JCUM)	P4712690	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
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0.0 TO TO THE PROPERTY IN	Px112170	*
	PMT12780	
243C2 (014.0) FFEE	PRT12790	
WRITE (6.170) (GMAS(I), I=1, NOPAN)	PRT12800	14004
670 CONFINIE	PRT12910	-
JLIMIT - NSOFAR	PRT12829	この、 は、
ACASM1 = 4SDFAR	PRT12830	
KRITE (6+220) KCNSAL	PKT12840	
MAILE (4.170) (411).1=1.1[M(f)	04117450	-
	PRT 12850	1
MRSTE (6,179) (EVILL, 1,1,1,14)	PR112970	DEAL PARTY IN CREATE IN 1. NO.
MRITE 46.2403 NC4S#1	PR112880	- 2
MRITE (6.173) (21(11.1*1.1.1.1 11)	PRILZAGO	
ARTIC (5.250) NCASAL	PR112900	
WRITE (6,170) (22(1),1-1,3_141T)	P4112910	-
MENTE (4.230+ NCNSE)	PRT12923	はしま アー・ こと かくまかかり シャイチョン・・・・・・・・・・・・・・・・・・・・・・・・・・・・・・・・・・・・
WRITE 16,170) (Y(1), (*1, JL[M[7])	PR112910	# 1
ERGITH (0.700) NCNSEL	P4T12940	C TOOL OF THE STATE OF
MPITE (6,173) (PV(11,1-1,JLEMIT)	PRT12950	C C I NAVARAN CC UN CAR CAR AND AL
ER11E (6.260) NC4541	PH112360	
MRSTE (6-170) (PE(11-1-1-1-14))	PRT12970	
delte (6,270) MCMS#1	PRT12980	37355 - 137355
WRITE (6.170) (P241).1-1.1[MIT)	PAT12990	CONTROL CONTROL OF THE CONTROL OF TH
WATTE (6,130) MCNSML	94113000	
MRITE (6,170) (2211), [= 1, JLIMIT)	PRT 13010	
•	PH113020	000
16,1731	PRT13030	-
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SUBTOUT INEGENETERS, DELLICELY, MA, WSIRIP, GETA, IME, NK, MP, MS ARAY, LIRIGEBEDGIO

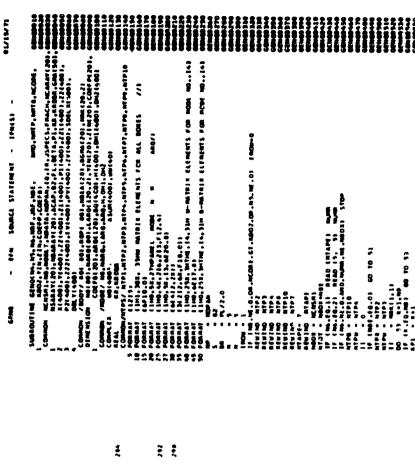
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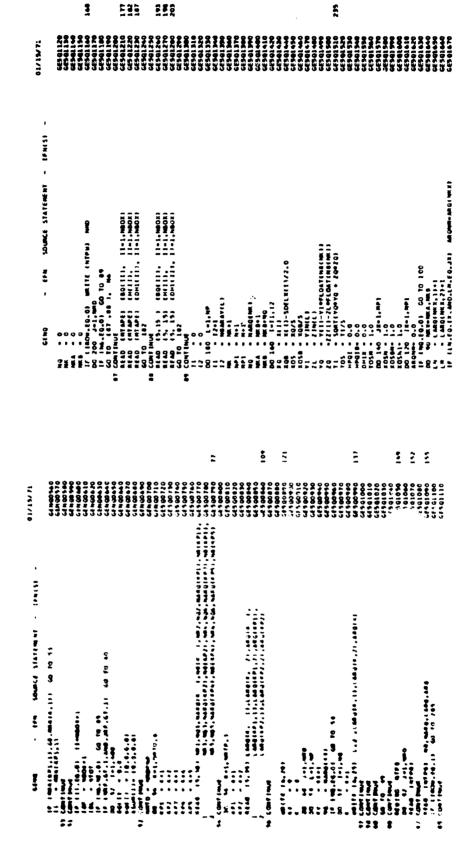
A MISCHIOSON CONTROL MAN TO A STATEMENT OF A 
                                      01/13/71
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- EFN SOURCE STATEMENT - 1FM(S) 22 . ton sounce statement . toniss THE STATE OF THE S 441 184 - 18 91 - GG TD 220 170-146116-189-200,400-4001- J 176-11682 N.CF (1.4) .18. 0.1 GO 70 400 506.31L) • 0ftv1L) INII, +1 . CAGEO MI. 11 . CACHO HE 1, 71 . CHOMO EMELLES - CACOR # 1-1 = 1 = 1 = 1 = 1 = 1 - 1.45 Ft 18 (18.57.18.5)
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100 DO 110 L - 1
10 E - L - 1 H 1158282 55

GENS - EFN SOUNCE STATEMENT - LFM(S) -	01/15/11	(S)M91 - [M983]F1S 930F0S M93 - M839	710
B418,1C) = 1D + DELTANIEVILI-ASIJOSDELXILIODELYLLI	CERRE TTO		
	GEN81 720	COSTA COSTA TOTAL DESCRIPTION OF STATE	1
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Ę	CENB1 760	CONTROL OF THE CONTRO	PO-14 PO-14 P1-04
### ##################################	GEN81770	LIME HALLOW LIME SOLD BUT HAS ILED TO SALANDE LE COLON OF ANY SOLD BUT AND ANY SOLD BUT ANY	1, Met. 21 450 1, 4
M. 4111 GT 441	GENB1 780	CONFLEX MENO, SONIC 4001	13
B(IR,IC) = (1001/001/050ELXIL)001/1	GENET 790	S FORMAT (BAL. 19K. 2 Pr. so & MATRIX ELEMENTS FOR THE. 14, 6M	. OH STRIPS, GO
GO TO 310	0101000	1 14, 104 BOXES 40//XX, 44/SXXIP MO.,3X,7XBOX MO., 3X,52X	22 = E.C. E.
SOUTHWEST THE STATE OF THE STAT	CEMB 1 820	TO FORMAT TANGET AND THE PARTY OF THE PARTY	7 4
DELTA- 11x9-x11 + (x9-x2)-/x01	GEN B 1 8 30	15 FORMAT (1M1,220%, 33H se GUST RIGHT-HAND-SIDES FOR THE 14, SH BOMESGE	.14. 4H BOXESG
_	CENS 1840	1//00 1	3
	GENB1840	I FURNAT I LAD. 26%. (ALCAMAR OF TAKE, 19). YA CYBORON	T a
32 50 50 1 • 1 30 1 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	GENB1870	30 FORMAT (100-25X-F10-6-1)	. 3
	044111	40 FORMAY (4F10.0)	3
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	01619895	OF THE POST OF	3 8
AND DESCRIPTIONS	GEN81920	CLEAR CRITISH	3 9
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16 (EVILL . 1 - 16) 60 10 150	GEN 1950	1F 1MGUST.EG.61 GO TO 80	•
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	CENBIARO		3
	GEN81940	TO THE TOTAL OF TH	3 8
	GENB 2000	CENTRAL PROPERTY OF THE PROPER	8 8
IF SEVIL), LT , M49 GO TO 370	6EN8 2010	CALS: NG-SERCENGE CELT CALS:	3
Offin, IC) . SOEL KELI POEL VIL	CENE 20 20	CREST = CLIMAN GRASS GRASS CONSTRUCT	3
37d CONTINUE	GENB2040	I CINCLING TO TO TO	31
	GEN82050	CETTE - CONTENTS - CON	5 4
	GENB2060	ERITE (6-30) FGAMMA(J)	7
IF (EVIL), AT .xe) CO TO 140	GEN82070	70 CDN TRUE	3
BITR. ICS - SDELMILIPOELVIL		MAINTE GA-191 NOON	3
SAGE CORE DELAN	GEN \$2 100	CALLY CONTROL OF THE CANCEN	3 0
	GEMB2110	80 CONT INUE	r 8
00 409 L + Linitalia 122	CENE 2 120	400 DQ 470 [m 1,857419	3
1 4 M17 - 1 - 27	GEN82140	0 - 32	31
AND AND A STREET	GEN82150		ra
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TO COMPLETE OF THE PROPERTY OF	GEN 8.2210	- QUOHO 001	
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CEMPORAL CENTRAL CEN		GENWO620	
The color The		GENNO630	MRE - DELTA
10 ++0		GEN#0640	
The control of the		GENEO650	280 MIR, IC) = CAPLX MRE, MIM)
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(X(L -X) CERNOTSO C	1 100 LT 00	GFMM0710	- 2005 - 2015 - 0.0-4M0. VII 1-61-X41
CENNO 750 CENN	7	CENTRO 240	- TOBBARDE
(#(L -#1)		GENEO 250	
-X11)		GENEO 760	2
-X31)	1.00	GEMM0770	: }
-X11)		GENNO 780	DELT
-X31)	IELO, KROBROCKIL I-XII)	GENED 790	•
-X31) GERMORD CONTINUE CONTINUE CERMORD STOLE LINE LAIN I GERMORD CERMORD STOLE LINE LAIN I GERMORD CERMORD STOLE LINE LAIN I LILL LAIN CERMORD CERMORD STOLE LINE LAIN CERMORD STOLE CERMORD CERMORD STOLE CERMORD CERMORD STOLE CERMORD CERMORD CERMORD STOLE CERMORD CERMORD STOLE CERMORD CERMORD CERMORD STOLE CERMORD CERMORD CERMORD STOLE CERMORD CERMORD CERMORD STOLE CERMORD CE		GENWO 800	•
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-531)	1	CP28C0820	
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CEMUCORO		GENWO360	TALE OF THE PARTY
CERNORSO	200	GENEGATO	
CERMONO CERM		GENNOBBO	
- 45) CFM0940 CFM0940 350 CFM0940 35	60 10 200	GENNOB90	IXI SLCT(1.41.NE. 0.0.4MD. X(L) .GT. X4)
CFRMC940 CFRMC940 S90 CFRMC940 CFR	11.0 KROSROCKIL I-XSII	GEMN0900	KEE 1 100 F / 60 1
CENNO 20		GENN0910	
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141 131 154 164 141	11102	GEN#1010	
014 001 (#W3)	-	SENW1080	if (x(t).LT.xe) 60 TO 410
01V	1) + [X(L)-13]	CE 1841 1040	
	2) • (x(L)-13)	001120	





GENG - EFN SOURCE STATEMENT - LFNIS) -	11/51/10		GENU - EFN SOUNCE STATEMENT - JENES) -	/\$1/13
	00010000		12 - MBABAYII k	
TOO COME TANDO OF	06201690		DO 280 1-11-12	200
HERE - TOTAL - CORPORATION - TOSA	GE501700		X X X X X	
•	GE5-01710		XO5 = NQ/S	20099
DMIX = DMIK - CJEFPILJOARQMOHDSMIFTOSMOFLGATIX-13	66501720		A VANCE	3
KO. 1 = KOSMeANOS	66501730			(FSQ2)
MC UNITE ROSENSES	GESQ1740			28630
021 01 02 07 07 07 07 07 07 07 07 07 07 07 07 07	001000		TO SERVICE ON	CE 502
CONTRACTOR AND	66501770		=	3
	SF501780			200
140 COOK INCH	06.201790		2G mf2C3b-2C+COAT(NBCNC1)	200
DELA: - SOFL #(1900EL 7(1)	GES Q1 800			
BOLLS = DELAL + MPGIB/(SeS)	06501910			205
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=	CE501830			CE 5023
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٠	GE592170		240 CONTINUE	CF50072
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	06.5000	3**	-	5630275
	0122200		275 CONTINUE	GE 302 76

	01/15/11	CTOSE "G-4CO" - T-4E" - 48K* 04 * 4K "PENK" PENK KESK "OFFICEOR SAFETIONEOR	0100000
CAME - LANGE STORE STATEMENT - LANGE -		CEMBON - MCMSWILL NOT	0.400m0020
		L NSAFAY(20) "NGAAAY(20)" AG AP. 87% F. BRYAR PE. KR. KR. KR. BRAGA-1100 BOCKON.	.AUG#0030
	074 07180 470	2 844001,**********************************	A GOMODAD
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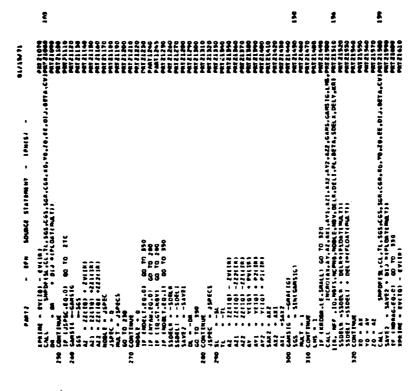
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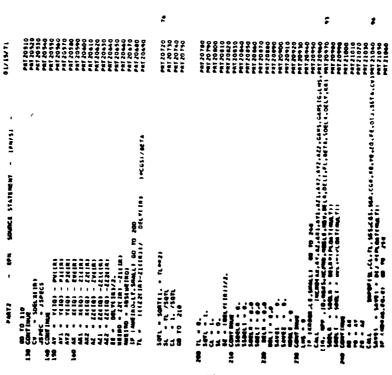
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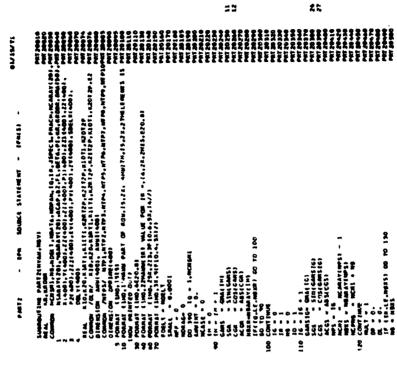
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- EFM SQUACE STATEMENT - IFNIS) .

PART 2

			14 . OF 341
7	E, ULJ, BETA, CVIPAT 21620	18	!
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330 CONTINUE	PR121840		CARACTER SAN TARGET
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340 GAMS16 v GMAR163	PRT 21670		45A: K10. L20. 61871
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AZ - 221601 - 24189	PRT 2 1 440		100411111111111111111111111111111111111
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41.134.134

 SOUNCE STATEMENT - IFMEST - EFN K1111 - 11 - CK1 K27872P - 129- CK2 K2712P - 129- CK2 K1011 - K10- 11 K2012P - K20- 129 CCNT (MUE END KERNEL 903 228 65 17 25 73 01/15/71 - EFN SOUNCE STATEMENT - IFNES! -

11.751.710

XERNEL

	01/11/11		
ine			10F2 - EFN SOLACE STATEMENT - 1FNISI -
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The state of the s	0000		FAC28= RISGN-ETAOLOEE
Š	10-10210		FAC38= R15GR-ETAO10EE
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1984 - 4851 4854)	10f 10230		TEMP214 (FACADAR21+FACADAR21+(ETAD1+EE)+C21)/OEX2
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	104 10 200		ARGA - CDEF+2ET01
大型 2番の まに	10610290		
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	9		CO TO 100
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1 - 1063001	10610370		ARGT = COEFOAZET
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			UP11 = 2.00(E20A21 + C21)

\$17.157.71 \$100.000 \$100.000 \$100.000

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SUBROUTINE CXSS (MCKSAI, MD, MCUST)

COMPLEX (MX 128.140), LIM 50.1101

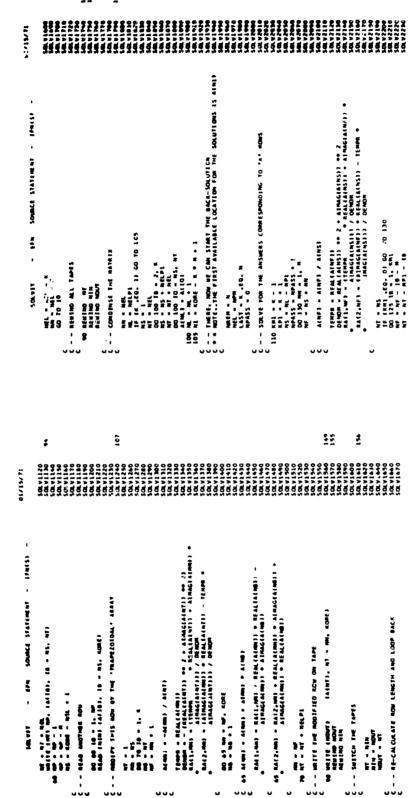
COMPLEX (MX 128.140)

COMPLEX (MX 12 - EFN SOUNCE STATEMENT - IFNES! 115 ? - EFN SOUNCE STATEMENT - IFMES) -| 120 CONTINUE
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| California | Cal C -- er 15 GREATER THAN 11 SQ ME CAN START THE TRIANGULARIZATION - EFN SOURCE STATEMENT - 1FM(S) RACL, MAI - RACL, MAI + REAL LACKTI) - REALLACMDI) --- CHECK TO SEE IF WE WERE UNLUCAY ENDUGH - - WEITE THE "TRAPEZOIDAL" MATREE ON TAPE - - FORM THE "TRAPEZOIDAL" ARRAY (B) 00 40 EB = 1, K MS = MT + 1, MT = MT + NEL 40 READ (MIN) (A(10), 10 = NS, NT) SO AIMMI - ACHNI + ACHTI + ACHSI 1F 4K .Eq. 11 GO TO 90 ACHT! - -ACHT! / ACHS! 17 114571 60 10 90 NELP! * NEL * ! NS = - NEL NELP? = NELP! * ! DO 50 10 - 2, K NP = MELP2 - 10 NS - 85 - 8EEP1 NS - 85 DC 50 10 - 10, K MS - NT - NE. NB - NS 00 50 MF - 2. #F SOLVIT 01/15/71 and I tuning Is, 74 E IS, 15s marate excess is, 7s womes, 1 consistent and the constant of the - EFN SOUNCE STATEMENT - IFNIS) -MITTER BY J. L. MESS & PROGRAMMED BY T. H. RIDGELL SUCRECTIFIE SOLVETING, NG, NE, NE, NG, NE, NPRES - FILLER 131818 CACACITA MAINE 4. 9 100: va. 00CE (att). 0att. 11) - - 1111 PO 100 10 PO 1111 G CONT. 14 -680) SINCELIES 14(1, -660) The / 1 The - Delivi - 1 91 ************* BICK LAST 71145

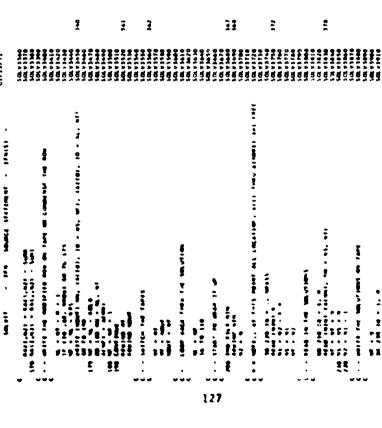


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01/15/71





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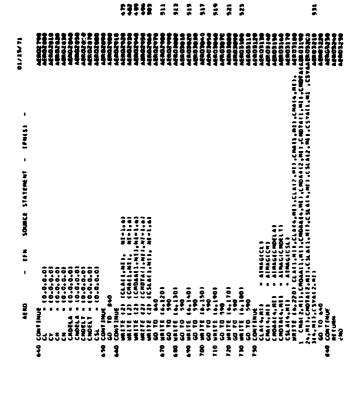
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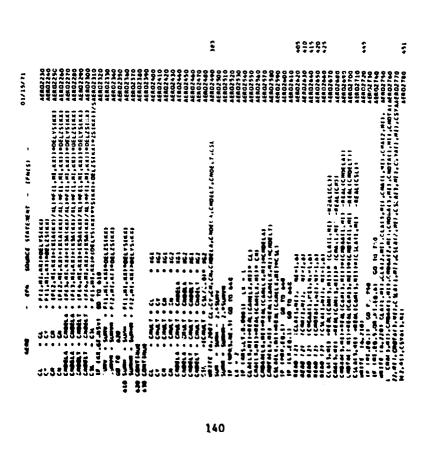
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Two methods of accounting for body-lifting surface interference in unsteady flow are considered. The first method is described in Part I of this report, while the second will be described in Part II to follow.

The first method is a direct application of nonplanar lifting surface elements to both the lifting surfaces and the body surfaces. The body is treated as an annular wing. This type of idealization must be used with an axial doublet introduced to account for body incidence effects. The undesirable effects of the annular wing representation are then reduced.

The second approach, to be described in Part II, uses an image system and an axial singularity system to account for the effects of the bodies.

This report also describes an improvement of the Doublet-Lattice Method of Albano and Rodden. The improvement pertains to wing-tail problems where there exists a small vertical (non-zero) separation between the wing and tail planes. Such problems can now be handled with ease.

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